

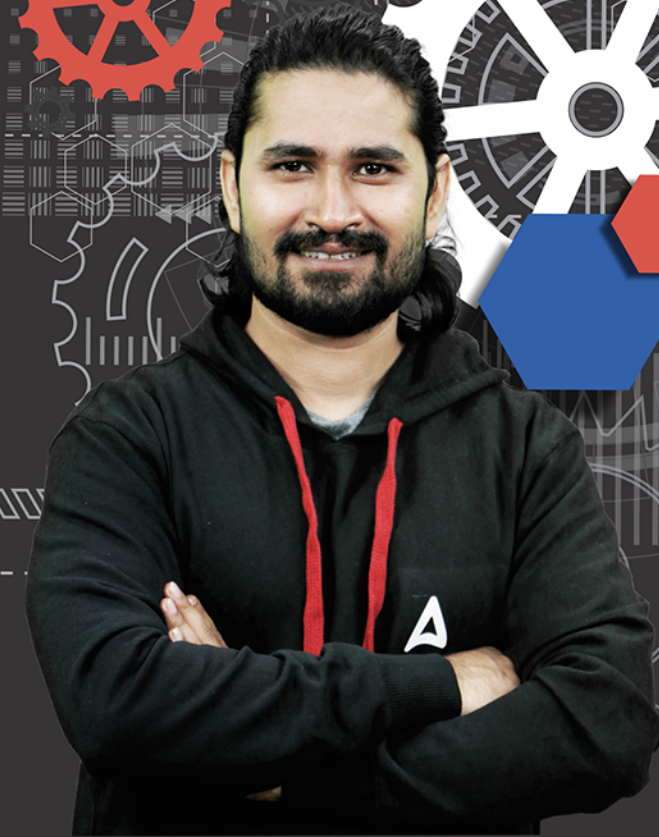
Complete Hand Written Notes

MECHANICAL ENGINEERING

FOR ALL AE/JE & PSU EXAMS

- › More than **750+ Pages** of hand written study notes
- › Tricks and Tips to remember Concepts
- › Crystal Clear Conceptual Explanation
- › Diagrammatic Representations of concepts for better understanding
- › Trend Analysis of Questions and many more

Shivam Gupta



Preface

Dear Engineering Enthusiasts,

Welcome to my handwritten book on mechanical engineering! I'm excited to invite you to explore the result of my hard work, passion, and Steady commitment. This book represents countless hours of crafting every concepts in words, diagrams, and equations to bring the essence of this field to real life.

Throughout this journey, I submerged myself in the complex world of gears, engines, and mechanical systems. My goal was to make complex concepts approachable and engaging, which is why I've written this book using four different colours to help you understand the concepts more easily.

To enhance your understanding, I've included figures that will provide visual clarity. This book aims to be your one-stop solution for clearing all AE/JE exams. You won't need any other book because it covers all the content you need.

In addition, I've conducted a thorough analysis of the last 10 years' subject-wise papers, and Set of 1000 practice questions in e-form in Adda247 app giving you a better approach to exam preparation. I've also taken care to design this book with simplicity and creativity in mind.

With sincerity, I present this handwritten book to you, hoping it becomes a cherished companion on your journey to becoming a AE/JE officers. Let it ignite your curiosity, passion, motivation and guide you towards realizing your dreams.

Author
Shivam Gupta

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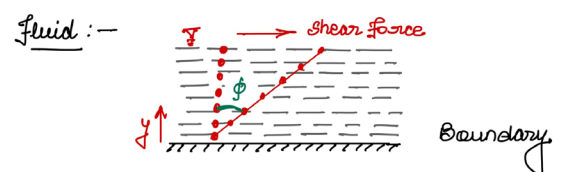


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Fluid Mechanics

Fluid Mechanics

Introduction to fluid and Properties



under no slip condition

$\frac{d\theta}{dy} \rightarrow$ Angular deformation } Due to intermolecular bonding
 $\frac{du}{dy} \rightarrow$ Velocity gradient }

- # The rate of deformation is important than deformation. In Solids, deformation is constant
- # Solid can resist any load, but fluid can resist only compressive load.

Fluid is a substance that is having the ability to flow under the action of shear force.

If shear force = 0 (Static fluid)

Ideal fluid : Zero viscosity / non-viscous / inviscid
 Zero surface tension
 Perfectly incompressible

"We assume water as an ideal fluid"
 In short it is भगवान जी / God fluid."

concept of continuum breaks down under :-

- # Shock
- # high vacuum condition
- # Rarefied gas flow
- Rocket \rightarrow flights \rightarrow upper atmosphere

Properties of fluid :-

Density / mass density / specific mass $\rightarrow \rho$

$$\left. \begin{matrix} \text{Pressure } \propto \text{Density} \\ \text{Temperature } \propto \frac{1}{\text{Density}} \end{matrix} \right\} \rho = \frac{\text{mass}}{\text{Volume}} = \frac{M}{V} = \frac{\text{kg}}{\text{m}^3} \text{ or } \frac{\text{g}}{\text{cm}^3}$$

[ML⁻³]

Analysis

We can define density in terms of

- # Escaping tendency of molecules
- # NO. of molecules / volume
- # Intermolecular force of attraction
- # Degree of freedom / movement
- # heaviness of the substance

** $\rho_{H_2O} = 1000 \text{ kg/m}^3$ | $\rho_{Hg} = 13.6 \times 10^3 \text{ kg/m}^3$ **
 $\rho_{air} = 1.2 \text{ kg/m}^3$ $\rho_{petrol} = 750 \text{ kg/m}^3$ Approx.

Specific weight / weight density " w " $w = \rho g$

$$w = \frac{\text{weight}}{\text{Volume}} = \frac{m \times g}{V} = Sg = \frac{N}{m^3}$$

weight = $S \times g \times V$ $[ML^{-2}T^{-2}]$

Specific volume $\rightarrow (1/\rho)$ Reciprocal of mass density

$$v = \frac{m^3}{kg} \quad [M^{-1}L^3]$$

Specific gravity / Relative density (S)

$$S = \frac{\rho_{\text{unknown fluid}}}{\rho_{\text{standard fluid}}}$$

Dimensionless quantity

$$S_{Hg} = 13.6$$

heavy

$$S_{H_2O} = 1$$

Water

$$S_{\text{petrol}} = 0.75$$

lighter

Liquid \rightarrow standard fluid \rightarrow water

Gas \rightarrow standard fluid \rightarrow Air or H_2

Compressibility & Bulk modulus



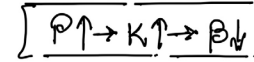
$$\beta = \frac{1}{K} = -\frac{dV}{V dP} = \frac{dP}{\rho dP}$$

Bulk modulus

$$K = \frac{\text{Volumetric / hydrostatic stress}}{\text{Volumetric strain}}$$

$$K = \frac{dP}{-\frac{dV}{V}} = -\frac{V dP}{dV} \text{ or } \frac{\rho dP}{d\rho}$$

$$K_{\text{(adiabatic)}} = \gamma \cdot K_{\text{(isothermal)}}$$



$$K_{\text{air}} = 0.103 \text{ MPa}$$

$$K_{H_2O} = 2060 \text{ MPa}$$

$$K_{\text{steel}} = 206000 \text{ MPa}$$

We treat

Air (static) \rightarrow compressible

Air (flowing) \rightarrow incompressible

Air is 20000 times compressible than water

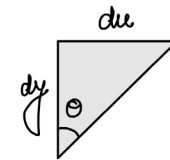
Water is 100 times compressible than steel.



(Although it is incompressible)

Viscosity

"Resistance to movement of one layer of fluid to the adjacent layers."



$$\tan \theta = \frac{du}{dy} = \frac{d\theta}{dt}$$

or
Measure of Resistance of fluid which is being deformed by either shear stress or tension.

$$\mu \propto \frac{du}{dy}$$

" μ " is constant with respect to $\frac{d\theta}{dt}$

$$\tau = \mu \frac{du}{dy}$$

Newton's law of viscosity,

Coefficient of dynamic viscosity,

Shear force $\leftarrow F = \frac{\mu \cdot A \cdot V}{y}$ \rightarrow Very less

$\mu = \frac{N \cdot s}{m^2} = Pa \cdot s = \frac{kg \cdot f \cdot sec}{m^2}$ \rightarrow MKS unit $[ML^{-1}T^{-1}]$

Constant of Proportionality or Absolute dynamic viscosity $1 \text{ Poise} = 0.1 \frac{Ns}{m^2}$

CGS unit $1 \text{ centipoise} = \frac{1}{100} \text{ Poise}$

Kinematic viscosity $\rightarrow \nu = \frac{\mu}{\rho}$ $\frac{m^2}{sec}$ Also known as Stoke's viscosity

$\nu = \frac{\mu}{\rho} = \frac{\text{dynamic viscosity}}{\text{mass density}} = \text{Resistance to molecular collision}$

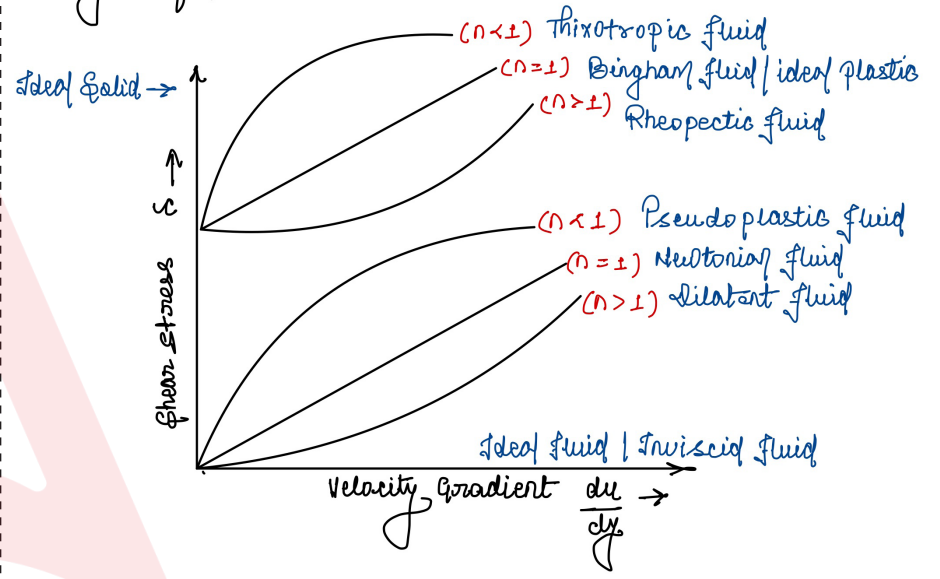
unit CGS: $1 \text{ stoke} = \frac{cm^2}{sec} = \left(\frac{1}{100}\right)^2 \frac{m^2}{sec} = 10^{-4} \frac{m^2}{sec}$
 $1 \text{ centi-stoke} = \frac{1}{100} \text{ stoke}$
 SI unit = MKS unit = $\frac{m^2}{sec}$ $[L^2T^{-1}]$

"Newton's law of viscosity" :- Shear stress \propto Rate of shear strain

Velocity Gradient \downarrow $\left[\tau = \mu \frac{du}{dy} \right]$
 $\frac{du}{dy} \rightarrow$ Rate of shear strain / Rate of shear deformation
 Rate of angular deformation

cohesive forces decreases
 $T \uparrow \rightarrow$ Liquid $\rightarrow \mu \downarrow$ (Intermolecular bonding)
 $T \uparrow \rightarrow$ Gas $\rightarrow \mu \uparrow$ (Molecular momentum)

Types of fluid



For non-newtonian fluids It obeys Power law $\tau = \tau_0 + K_s \left(\frac{du}{dy} \right)^n$

τ_0 = Shear yield stress
 K_s = Consistency index ; considered as " μ " in practical
 n = flow behaviour index

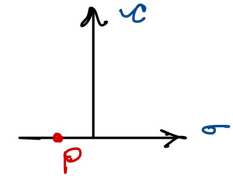
Dynamic viscosity	$ML^{-1}T^{-1}$
Power	ML^2T^{-3}
Moment of momentum	ML^2T^{-1}
Volume modulus of elasticity	$ML^{-1}T^{-2}$
Kinematic viscosity	$M^0L^2T^{-1}$
Surface tension	ML^0T^{-2}

- # Boiling is non-spontaneous process
- # Vaporisation is spontaneous process.

Fluid Statics :- (Pressure and its measurement)

$$P = \frac{\text{Normal Force}}{\text{Area}}$$

Zero pressure = No molecules
i.e.

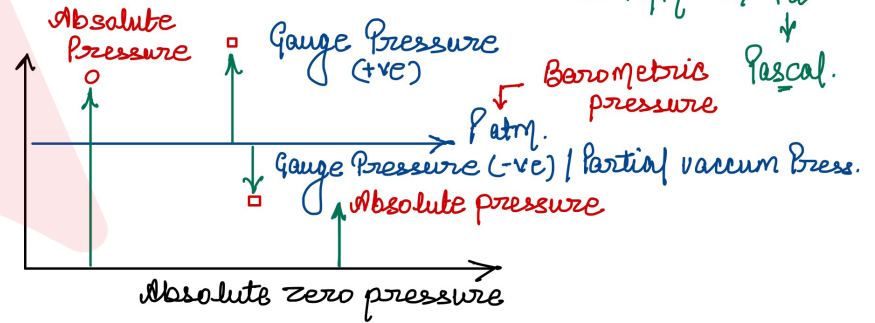


pressure corresponds to → molecules → momentum

Pressure ← forces @ boundary

$$\begin{aligned}
 P_{atm} &= 1.013 \text{ bar} \\
 &= 101.325 \text{ kPa} \\
 &= 760 \text{ mm of Hg} \\
 &= 10.3 \text{ m of H}_2\text{O}
 \end{aligned}$$

$$\begin{aligned}
 P &= \frac{F}{A} = \frac{kg F}{m^2} \text{ (MKS)} \\
 &= N/m^2 \Rightarrow \text{MPa} \\
 &= N/m^2 \Rightarrow \text{Pa} \\
 &\quad \downarrow \\
 &\quad \text{Pascal.}
 \end{aligned}$$



$$1 \text{ Torr} = 1 \text{ mm of Hg} = \frac{1}{760} \text{ atm} = 133 \text{ Pa} = 1.33 \text{ bar}$$

Evanzellista torricelli → Inventor of BAROMETER

Barometer is used to measure the atmospheric pressure.

Point to Remember



Hydraulic Machines

flow through jets

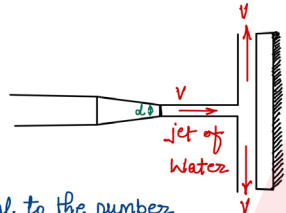
stream → Impact load
To rotate the turbine blades

Jet can travel the long distance without dissipating the energy.

Force exerted by the fluid jet on stationary plate

* When plate is normal to the jet

v → velocity of jet
 d → dia of jet
 a → area of x-s/c of jet = $\frac{\pi}{4}d^2$



Force exerted by the jet normal to the number

$$F_n = \rho a v^2$$

a = Area of jet
 v = Velocity of jet

Inclined plate

$$F_n = \rho a v^2 \sin \theta$$

$$\frac{Q_1}{Q_2} = \frac{1 + \cos \theta}{1 - \cos \theta}$$

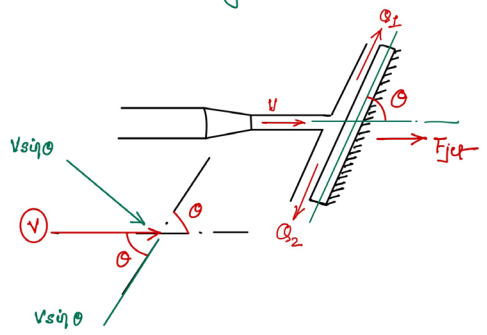
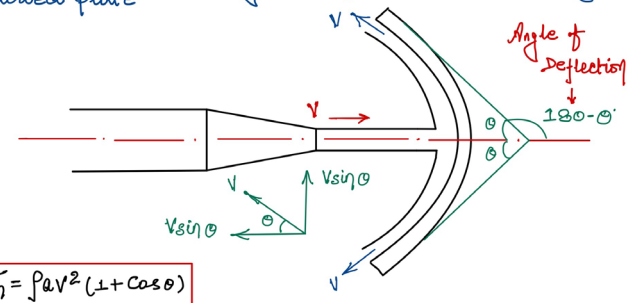


Plate is curved

Jet striking on a symmetrical stationary curved plate

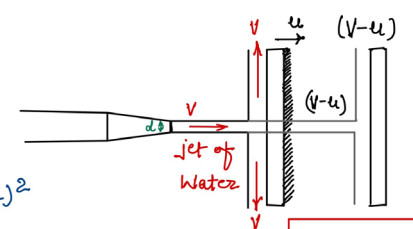


$$F_n = \rho a v^2 (1 + \cos \theta)$$

Force exerted by jet on a curved vane is always greater than that exerted on a flat plate.

Force Exerted by jet on a moving plate

plate is normal to the jet



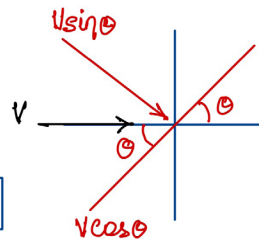
$$F_n = \rho a (v-u)^2$$

Work done per second (W) = $F_n \times u = \rho a [v-u]^2 \times u$

Jet on Inclined plate

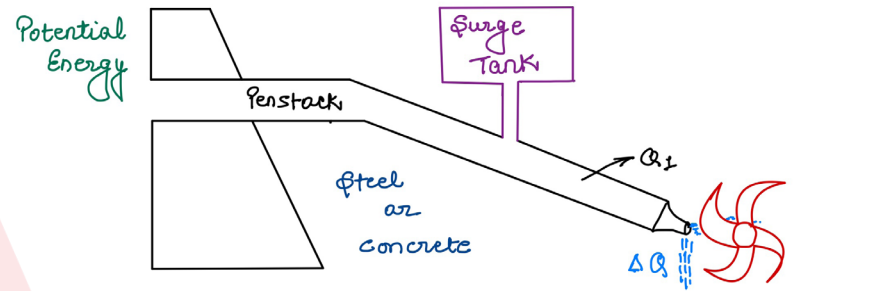
$$F_n = \rho a (v-u)^2 \sin \theta$$

$$F_{jet} = \rho a (v-u)^2 \sin \theta \cos \theta$$



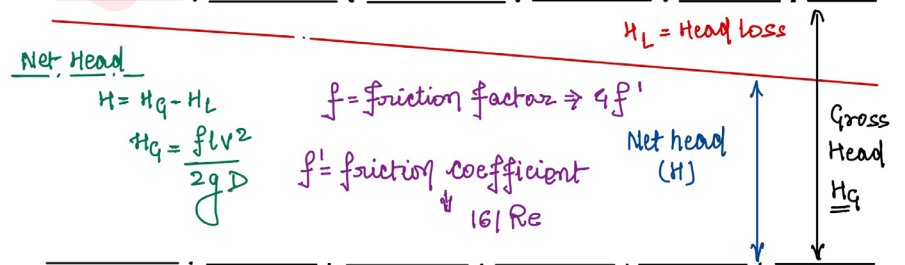
Hydraulic Turbines

"Rotational mechanical energy \rightarrow Electrical energy"



Surge tank :- It is small storage tank, which acts as safety tank for penstock against water hammer.

Different types of head



Point to Remember

”



“

CHAPTER
03

Strength of Material (Som)

Strength of Materials

Mechanics :- Study of motion, forces & their effects



Mechanics

Engg Mechanics - For Rigid body

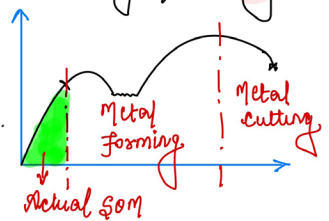


$f \neq 0, S = 0$
 $f = Kx$
 $K = \frac{f}{x} \rightarrow \infty$ or Zero deformation

SOM / MOM / MOS / Solid mechanics :- "Theory of elasticity"

study of deformable bodies
Deformation $\neq 0$

Mechanical engineering by dhruv Sir.

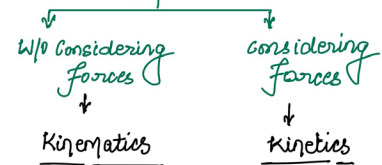


Fluid mechanics \rightarrow Study of flowable bodies



Engg mechanics

- ↳ Static - Rigid body at Rest
- ↳ Dynamics - Rigid body is in motion

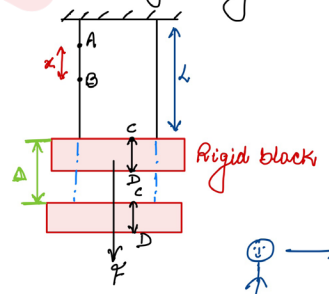


We study this with the help of

- * displacement $\rightarrow S$
- * velocity $\rightarrow ds/dt$
- * acceleration $\rightarrow dv/dt$
- * jerk $\rightarrow da/dt$

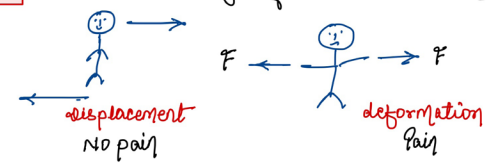
↳ Newton's 2nd Law $\rightarrow M \times a$

Engineering mechanics Vs SOM



$l \rightarrow l + \Delta$
 $x' \rightarrow x + S \rightarrow$ deformation
 $CD \rightarrow CD$
 No Deformation in block

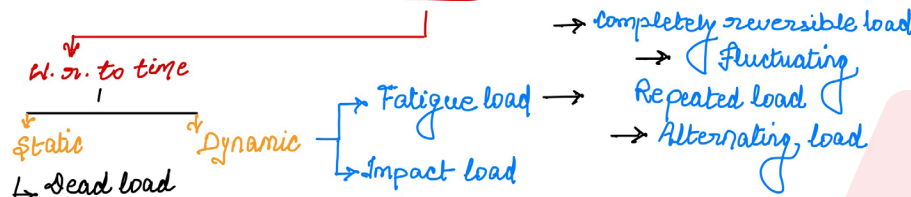
Disp of block = $\Delta \rightarrow$ Engg Mechanics
Defo. of wire = $\Delta \rightarrow$ SOM



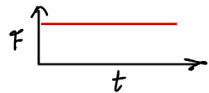
Loads & their effect

can be anything but
According to SOM
load is force or couple
moment is effect of force/couple

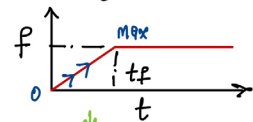
LOAD



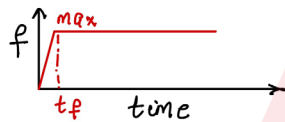
Sofa on the floor



Gradually applied load



Impact load



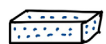
t_f → time required to become maximum

Impact $t_f < T/2$

Q.A.L. → $t_f > 2T$

It is coming under static load because the rate of deformation is very small.

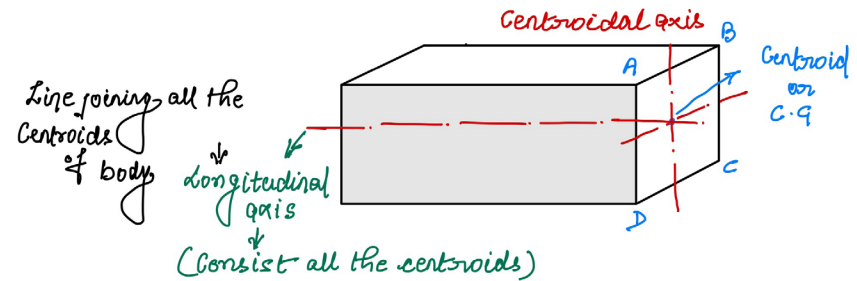
T = time period for Phononical vibration



It will have no. of molecules which are keep on vibration at room temp

Fatigue load :- Either magnitude or load or both changes with respect to time.

Load → With respect to direction



W.r. to Direction

Perpendicular to x-s/c
|| to Long. axis
(Normal load)

Axial Ecc. Axial

|| to x-s/c &
Perpendicular to longitudinal axis
(Shear load)

Direct transverse shear load Eccentric TSL

Load w.r. to Area

Concentrated Point load

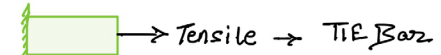
Point load Point moment

Distributed load

- UDL
- UVL
- UDM
- Sinusoidal
- UVM
- Variable loading

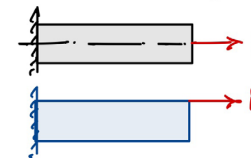
Types of Members

Axial loading



Vertical strut \Rightarrow Column

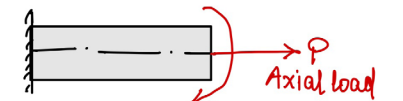
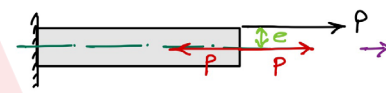
Equivalent effect of load



$\sigma = \frac{P}{A}$ [Formula known only for C.G. or centroid]

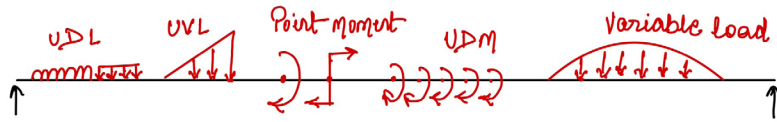
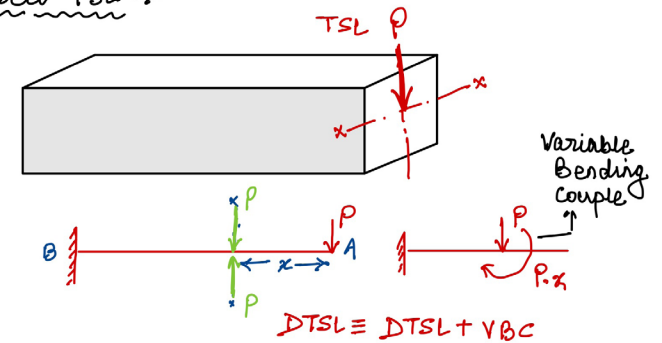
Here we need to see the equivalent effect of load

Eccentric axial loading:



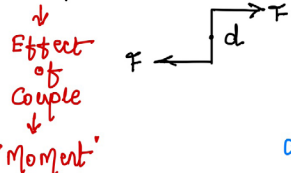
Ecc. A.L = Axial load + Bending couple
 (P) $(P.e)$

Effect of direct TSL \rightarrow

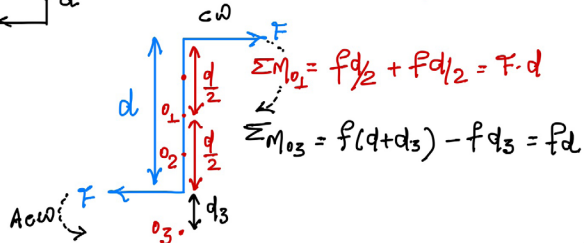


Vector

Couple $\rightarrow G = F \times d$



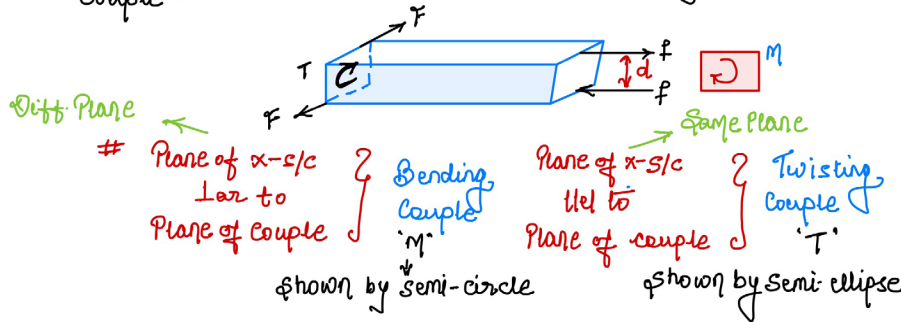
Effect of couple is independent of the point of Application

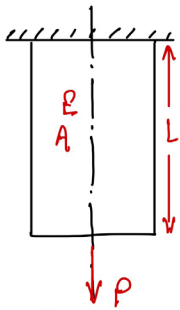


"Vectors whose effect is independent of the point of Application is said to be Force vector"

Thus

"Couple is a force vector" - IAS-2004



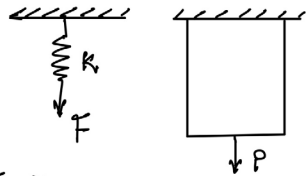


$$\sigma = \frac{P}{A} \quad \sigma = E \epsilon$$

$$\epsilon = \frac{\delta}{L} \quad \frac{P}{A} = E \frac{\delta}{L}$$

$$S = \frac{PL}{AE}$$

↳ The bar is prismatic through out
 ↳ loading is gradually applied load.



$$F = K \cdot x$$

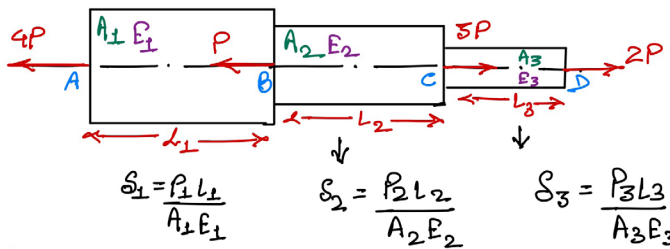
$$K = \frac{F}{x}$$

$$P = K_{bar} \cdot \delta$$

$$K_{bar} = \frac{P}{\delta} = \frac{AE}{L}$$

Axial Rigidity
↓
Stiffness of bar

Bars in series :- (Compound bars) → Both ends are free.



Individually it is correct but we need to find total Deformation.

So we can say

δ_1 = deformation for bar 1 & displacement for bar 2 & 3
 δ_2 = deformation for bar 2 & displacement for bar 3
 δ_3 = deformation for bar 3 only.

$$\delta_{total} = \delta_1 + \delta_2 + \delta_3$$

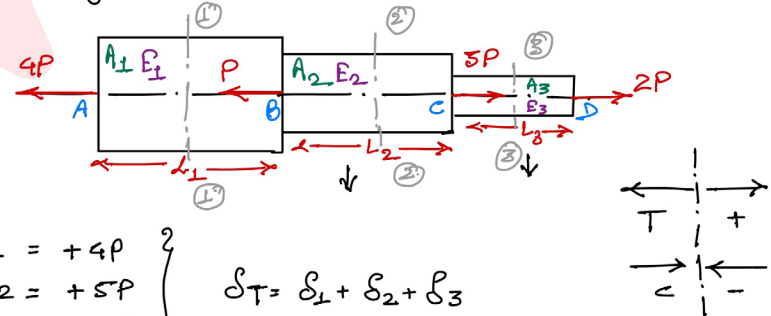
In series combinations, deformations are added up.
 to get $\delta_1, \delta_2, \delta_3 = P_1, P_2, P_3$

unknown

There are three methods to get P_1, P_2 & P_3

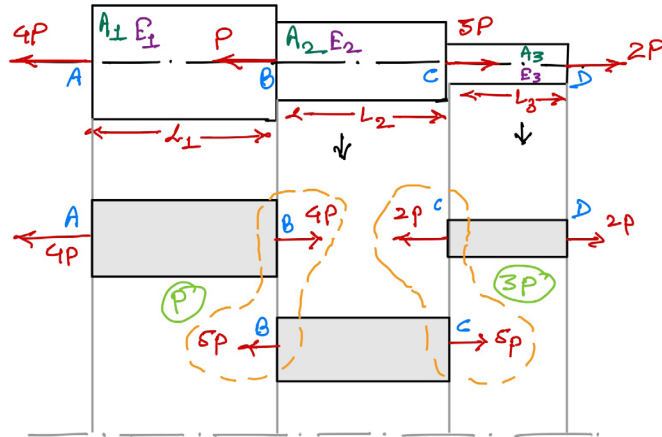
- 1:- Method of section
- 2:- Free body diagram
- 3:- Axial loading diagram.

Method of section :-

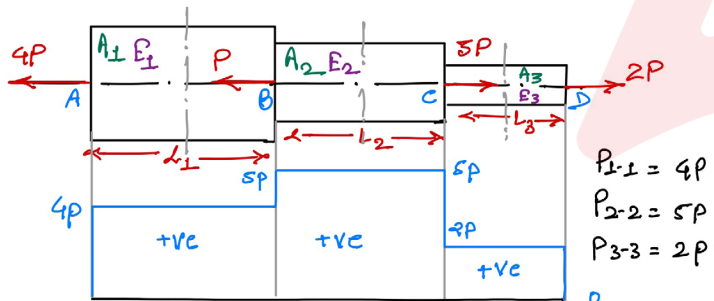


This method states force in a member is algebraic sum of all the forces either to the LHS or to the RHS of the section.

Free body diagram :-

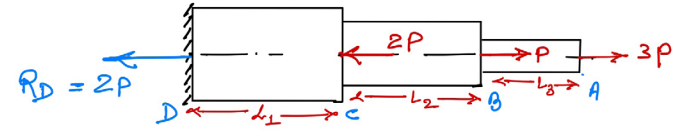


Axial Loading diagrams



Case-2 :- If one end is fixed.
Support → To give reaction

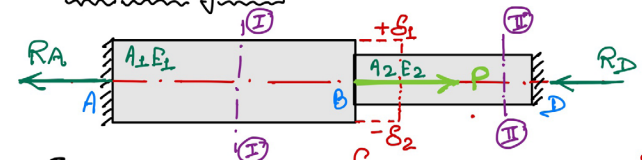
Total force will be zero because body is in equilibrium.



$$P_{1-1} = 2P \quad P_{2-2} = 4P \quad P_{3-3} = 3P$$

If one end is fixed, find out Support Reaction using equilibrium.
It will become → Both end free case.

Case → 3 → Both end fixed:



$$E_{Fx} = 0$$

$$-R_A + P - R_D = 0$$

$$R_A + R_D = P \quad \text{--- C1}$$

no. of eqⁿ = 1
no. of unknown = 2

Such condition

↓
Statically indeterminate

For such conditions we need special type of equation
"Compatibility eqⁿ"

$$\delta_T = 0 \quad [\text{Total Deformation}]$$

Put the value in equation 2 we will get $R_A = \dots$
then from eqⁿ 1; $R_D = \dots$
then $P_1 \neq P_2 \Rightarrow \delta_1 \neq \delta_2$

$$\delta_{AB} + \delta_{BD} = 0$$

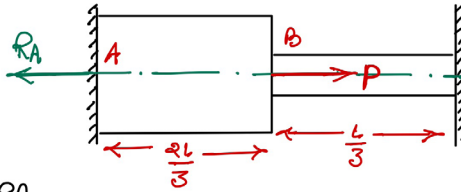
$$\delta_1 + \delta_2 = 0 \quad ; \quad \delta_1 = -\delta_2$$

$$\frac{P_1 L_1}{A_1 E_1} = -\frac{P_2 L_2}{A_2 E_2} \quad \text{--- C11}$$

$$P_1 = +R_A$$

$$P_2 = +R_A - P$$

Case-4 :- Both end fixed ($A_1 E_1 = A_2 E_2$)



$P_1 = RA$
 $P_2 = RA - P$

$RA \times \frac{2L}{3} = -(RA - P) \times \frac{L}{3}$

$2RA = -RA + P$

$RA = \frac{P}{3} ; RC = \frac{2P}{3}$

$RA + RC = P$

$\delta_1 + \delta_2 = 0$

$\delta_1 = -\delta_2$

$\frac{P_1 L_1}{A_1 E_1} = -\frac{P_2 L_2}{A_2 E_2}$

Shortcut :- Only for $A_1 E_1 = A_2 E_2$

$R = \pm \frac{\text{load} \times \text{distance of point of application of load from opposite end}}{\text{Total length}}$

$RA = \pm \frac{(P) \times \frac{L}{3}}{L} = \pm \frac{P}{3} = +\frac{P}{3}$
 $RC = \pm \frac{P \times \frac{2L}{3}}{L} = \pm \frac{2P}{3} = +\frac{2P}{3}$

+ \rightarrow opposite dirn
- \rightarrow same direction

Stress :

Internal resistive force developed at a point, stress and strain 2nd order Tensor.

$\sigma_{avg} = \frac{F}{A} \text{ or } \frac{P}{A}$

Internal resistive force offered by α -s/c of member.

Strength :- maximum/limiting value of stress that a material can withstand.

for mild-steel (ductile)

σ_{yt} (yielding strength) : 250 MPa

σ_{ut} (ultimate strength) : 400-450 MPa

for cast-iron (Brittle)

$\sigma_{ut} = (600 \text{ to } 700 \text{ MPa})$

Strength can be calculated by load-elongation curve.

Ductile materials are weak in shear.

Brittle materials are weak in tension.

$(\sigma_{total})_{ductile} \gg (\sigma_{total})_{brittle}$

State of plane stress occurs at the surface & state of plane strain occurs at the interior part of the plane.

Point to Remember




For Rectangle :- $\tau_{max} = \frac{3}{2} \tau_{avg}$

Solid Circular section :-



$\tau_{max} = \frac{4}{3} \tau_{avg}$

Triangular section :-

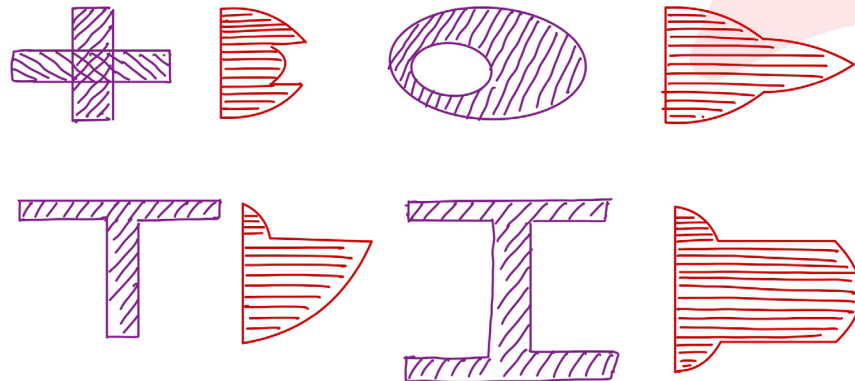


$\tau_{max} = \frac{3}{2} \tau_{avg}$

Diamond section :-



$\tau_{max} = \frac{9}{8} \tau_{avg}$



W rectangle < W square < W circle. W: Weight.

Torsional shear stress in shaft

$$\frac{T_R}{J} = \frac{\tau_{max}}{R \text{ or } R_0} = \frac{G\theta}{L}$$

for circular x- s/c.

Every power transmission shaft is not under pure torsion, but a part of a shaft is under pure torsion.

T_R → Resisting torque / moment of Resistance.
 T → Applied torque.

$T \leq T_R$ → Safe side.

τ_{max} → maximum torsional shear stress
 θ → max angle of twist
 L → length of the shaft
 G → shear modulus.

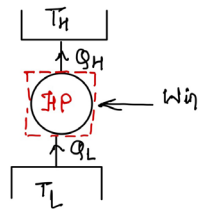
ϕ = max shear angle on the surface of the shaft

$$\phi = \frac{R\theta}{L}$$

Power = $P = \frac{2\pi NT}{60}$ → N-M.
 kW ← P , RPM ← N

Refrigeration & air-Conditioning (RAC)

Refrigeration & air-Conditioning



$$(COP)_{HP} = \frac{\text{Desired effect}}{\text{Work input}} = \frac{Q_H}{W_{in}}$$

$$(COP)_{HP} = \frac{Q_H}{Q_H - Q_L} = \frac{T_H}{T_H - T_L}$$

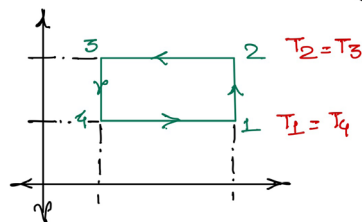
$$(COP)_R = \frac{Q_L}{W_{in}} = \frac{Q_L}{Q_H - Q_L} = \frac{T_L}{T_H - T_L}$$

$$(COP)_{HP} = 1 + (COP)_R = \frac{1}{\eta_{\text{engine}}}$$

The above expression is applicable only b/w the same temp limits.

Ideal Refrigeration cycle

"Reverse Carnot cycle"



- P(1-2): Isentropic or Reversible adiabatic compression
- P(2-3): Isothermal heat rejection
- P(3-4): Isentropic or Reversible adiabatic expansion
- P(4-1): Isothermal heat addition

$$COP = \frac{DE}{W_{in}} = (COP)_{RCC/IRC} = \frac{T_L}{T_H - T_L}$$

Reverse Carnot COP is the function of temp only.
Reverse Carnot COP is independent of working fluid.

Unit of Refrigeration : TR

$$1 \text{ TR} = 3.5 \text{ kW} = \frac{210 \text{ kJ}}{\text{min}} = \frac{50 \text{ kcal}}{\text{min}}$$

TON of Refrigeration = TR

Producing ice @ 0°C

$$(COP)_{\text{Summer}} < (COP)_{\text{Winter}}$$

because

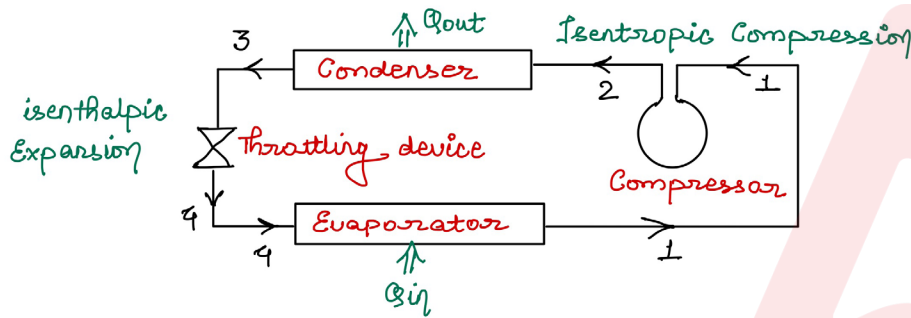
$$(T_H)_S > (T_H)_W$$

$$(T_{cr})_{H_2O} = 374.15^\circ C$$

$$P_{cr} = 221.2 \text{ bar}$$

Do not apply $PV = mRT$ in wet region

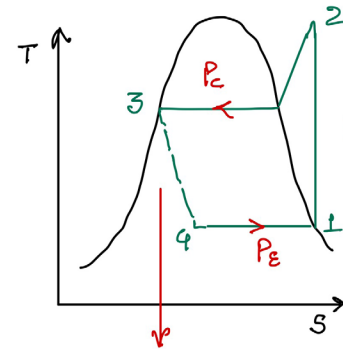
VCRS: Vapour compression Refrigeration System



Entry to compressor / Exit to evaporator
= Saturated vapour

Exit to condenser / Entry to throttling
= Saturated liquid

$$COP = \frac{DE}{W_{in}} = \frac{h_1 - h_4}{h_2 - h_1}$$



P(1-2): Isentropic Comp.
P(2-3): Heat rejection @ $P=c$
P(3-4): Isenthalpic expansion
P(4-1): Heat addition @ $P=c$

Irreversible process Represented by dotted lines.

Refrigeration Effect

$$RE = h_1 - h_4$$

KJ/Kg

Refrigeration capacity

$$= \dot{m} \times RE$$

$KW = KJ/Sec$

Work Input

$$w_{in} = h_2 - h_1$$

KJ/Kg

Power Input

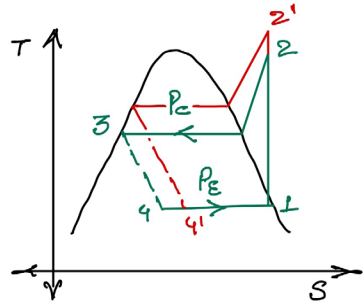
$$= \dot{m} \times w_{in}$$

KW

Point to Remember

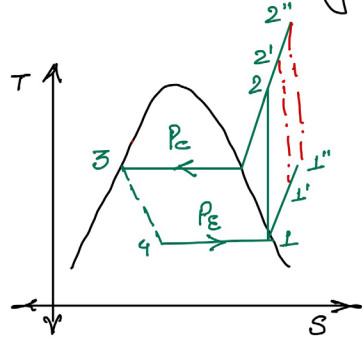


Case-2: ↑ in condenser pressure



$R_E \downarrow$
 $W_{in} \uparrow$
 $COP \downarrow = R_E \downarrow / W_{in} \uparrow$
 $\left(\frac{P_c \uparrow}{P_e}\right) \uparrow = \eta_{vol} \downarrow$

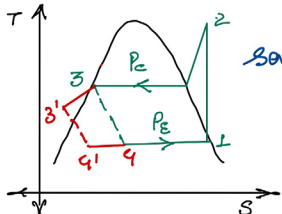
Case 3: Superheating



$R_E \uparrow = h_1' - h_4$
 $W_{in} \uparrow = h_2' - h_1'$
 $COP = \text{Can't say } \uparrow \text{ or } \downarrow$

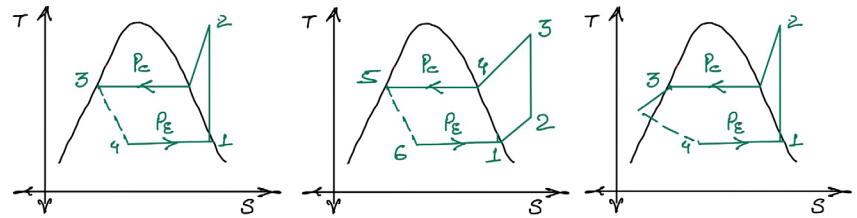
Work input to the compressors ↑ because it is a function of inlet temp of compressor.

Case 4: Sub-cooling: it is a process of ↓ temp. at constant pressure below saturated liquid.



$R_E \uparrow$, W_{in} : No change, $COP \uparrow = \frac{R_E \uparrow}{W_{in}}$

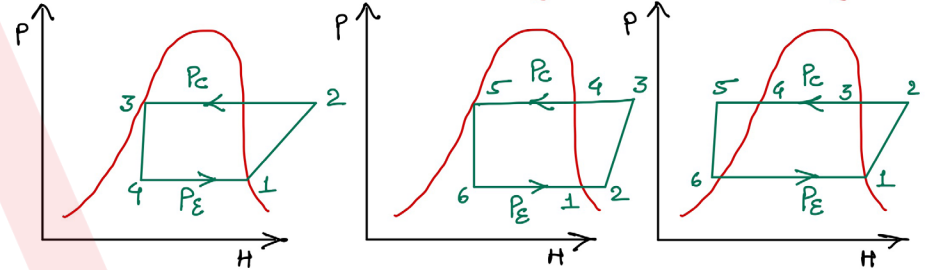
VCRS [T-s diagram to P-h diagram]



Simple VCRS

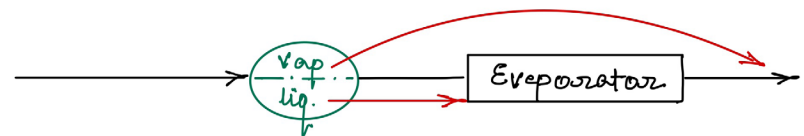
Superheating

Sub-cooling



Use of flash chambers in VCRS

It is a device used to separate liquid refrigerant from the mix. of refrigerant. it allows only the liquid refrigerant to enter into evaporator.

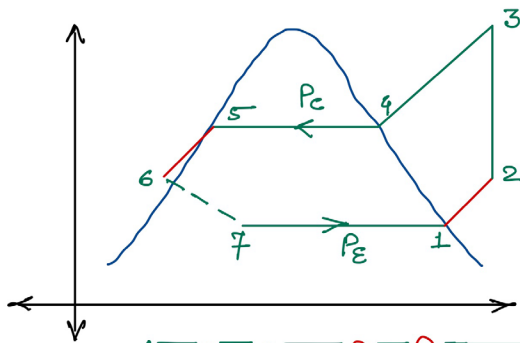


There is no impact on COP, with the use of flash chamber, it helps in reducing the size of evaporator.

Cascade Refrigeration system

$$(COP)_{cc} = \frac{(COP)_1 \times (COP)_2}{1 + (COP)_1 + (COP)_2}$$

Use of Heat-exchanger in VCRS



$R_E \rightarrow h_1 - h_7$
 $W_{in} \rightarrow h_3 - h_2$
 $Q_c \rightarrow h_5 - h_8$
 $h_6 = h_7$

$$h_2 - h_1 = h_5 - h_6$$

$$C_{p,v} (T_2 - T_1) = C_{p,l} (T_5 - T_6)$$

Refrigerants

Primary

R-11, R-12, R-22

Secondary

H₂O, Brine (NaCl solution)

Primary, which cools directly and secondary which is cooled by primary and then used for cooling.

Designation of Refrigerants

Case 1 :- When Refrigerant is saturated Hydro Carbon

$R - (m-1)(n+1)P$
 $n+p+q = 2m+2$



- R-11 \rightarrow CFCl₃
- R-12 \rightarrow CFCl₂
- R-22 \rightarrow CHF₂Cl
- R-134 \rightarrow C₂H₂F₄Cl₀

R-134 / R-134a

Ecofriendly Refrigerant because of absence of 'cl' element

Point to Remember



← 'cl' element deplete the thickness of ozone layer.

Case-2 :- When Refrigerant is unsaturated hydrocarbons.

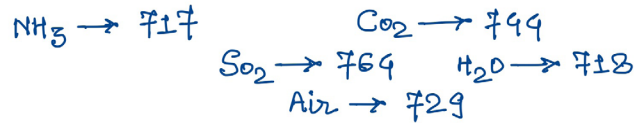
$$R - (m-1)(n+1)P$$

$$n+p+q = 2m$$



Case-3 :- When Refrigerant is organic compound

$$R - (700 + \text{molecular weight})$$



Desirable properties of Refrigerant

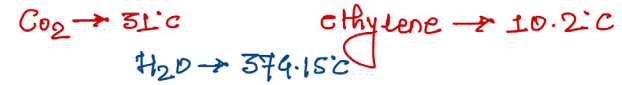
(Selection of Refrigerant)

Thermodynamic property

a) Critical temp :- Should be high or atleast above the condenser pressure

& temp

The C.T. of CO_2 & ethylene are almost undesirable for the Indian summer ambient condition.



b) Specific heat :-

Vapour :- High
Liquid :- Low

Enthalpy of vaporization as high as possible, it reduces mass flow rate \dot{m} .

Among the commonly used Refrigerants **AMMONIA** has higher value of enthalpy of vaporization.

$$R_c = m \downarrow \times h_{fg} \uparrow$$

(d) conductivity :- as high as possible

$$k \propto \frac{1}{\text{Area}}$$

e) Compression Ratio :-

low comp. ratio : \downarrow Winput : \uparrow η_{vol} .

f) freezing Point :-

as low as possible

h) Compressor discharge temp :-

$NH_3 \rightarrow$ Higher Comp. discharge temp \rightarrow Water cooled

Ex:- Freon leaks, Halocarbon detection
 ↳ Halide torch test
 blue to bluish green
 ↳ Soap bubble method

Ammonia leak detection
 Sulphur sticks method
 white fumes of ammonium sulphide is formed
 SO₂ leaks are detected by ammonia swab test

Refrigerant & their Application:-

- R-11 → large central AC-plant
- R-12 → Domestic refrigerator, water cooler
- R-22 → Window A-c
- NH₃ → Cold storage plants
- CO₂ → Direct contact freezing of foods
- Brine → Milk chilling plants
- Air → Gas liquefaction
 Air-craft Refrigeration system.

VRS → VARS

Vapour Absorption Refrigeration System

Compressor is replaced with G.P. Generator, Absorber, & pump.
 • COP of VARS is low 0.3 to 0.5
 • VARS is less noisy as compared to VCRS.

VARS is generally preferred in remote location where the cost of electricity is high.

NH₃ + H₂O → Ammonia → Refrigerant
 Water → Absorber

In order to remove the water particles from the ammonia vapours "Analyser and Rectifier" assembly is used.

The complete elimination of water particles occurs in Rectifier.

LiBr + H₂O → (Lithium bromide)
 Water → Refrigerant
 LiBr → Absorber

(The above pair is not preferable below zero 0°C)

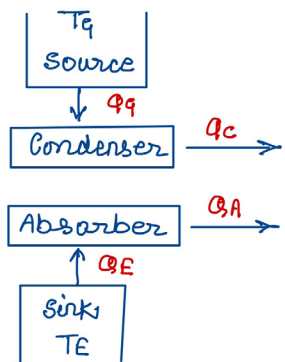
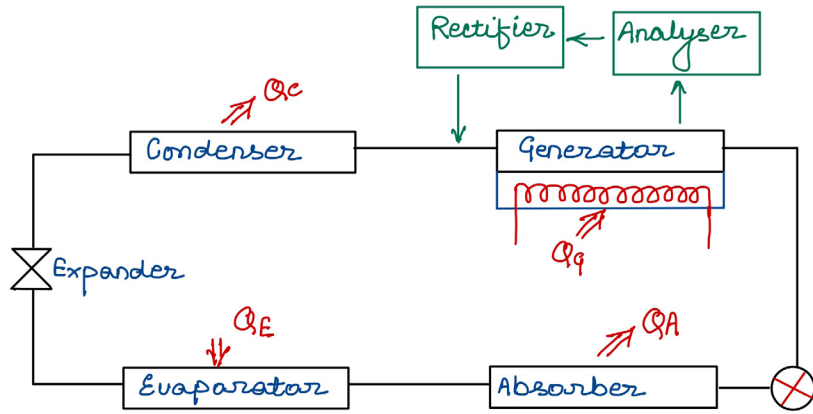
$$(COP)_{Actual} = \frac{Q_E}{W_P + Q_G}$$

↑
 pump work, it can be neglect also.

Point to Remember



66



$$(COP)_{VARS} = \frac{T_E}{T_g} \left(\frac{T_g - T_o}{T_o - T_E} \right)$$

$$\left. \begin{matrix} T_g \\ T_o \\ T_E \end{matrix} \right\} \begin{matrix} E \\ R \end{matrix} \quad \left(\frac{1 - \frac{T_o}{T_g}}{\frac{T_E}{T_o - T_E}} \right)$$

$$COP = \frac{T_E}{T_g} \left(\frac{T_g - T_o}{T_o - T_E} \right)$$

Electroflux generator

Noiseless operation

i.e. No use of pump

it is a three fluid system

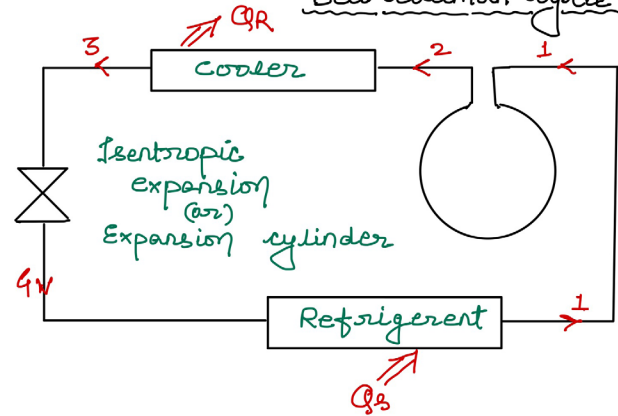
Ammonia : Refrigerant

Water : Absorbent

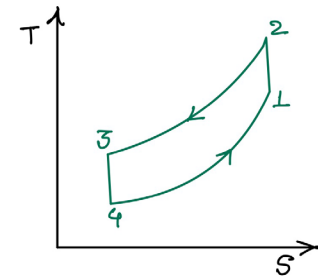
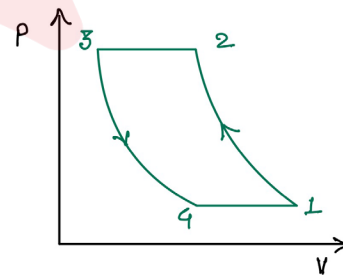
Hydrogen (H₂) : used to create low partial pressure of NH₃ vapour.

Reverse Brayton cycle

or Bell Coleman cycle



- Process 1-2 → Isentropic compression
- Process 2-3 → P=c heat rejection
- Process 3-4 → Isentropic expansion
- Process 4-1 → P=c heat addition



$$COP = \frac{1}{\frac{T_2}{T_1} - 1} = \frac{1}{(\gamma_p)^{\frac{\gamma-1}{\gamma}} - 1}$$

IC Engine

CHAPTER
08

IC-Engine

Heat engine : chemical energy of fuel : Thermal energy to produce mechanical work.

- | | |
|----------------|-----------------------|
| Cylinder | - Cast Iron |
| Piston | - Aluminium alloy |
| Piston Ring | - Silicon C.I. |
| Connecting rod | - Steel |
| Crank shaft | - Alloy steel |
| Bearing | - White metal |
| Cylinder liner | - Nickel alloy steel. |

Scavenging :- "process of cleaning" fresh fuel/mixture pushes the exhaust to exhaust port

4-Stroke engine

for the same power heavier engine is required.

more volumetric efficiency & thermal efficiency.

motor cycles, cars, buses, Trucks, aeroplanes, power generation.

2-Stroke engine

for the same power lighter engine is req.

less volumetric and thermal efficiency.

lawn mower, scooter, motor cycles, moped, ships.

otto cycle

Quantity governing
Lower efficiency
Lower compression Ratio (6 to 10.5)

otto cycle

Two constant volume & Two adiabatic process

Compression Ratio

$$\alpha_c = \frac{V_c + V_s}{V_c} = \frac{V_1}{V_2}$$

Mean effective pressure

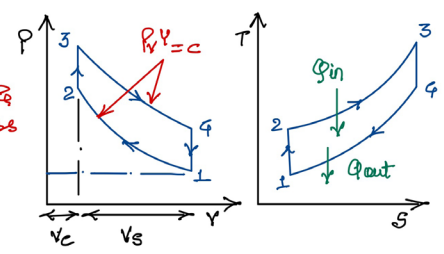
$$P_{mean} = \frac{\text{Work done}}{\text{Swept volume}}$$

$$(V_1 - V_2) P_{mean} = \text{Work done}$$

* As compression ratio \uparrow , efficiency \uparrow .
* Limit of compression ratio is subjected to knocking phenomenon is SI-engine.

diesel cycle

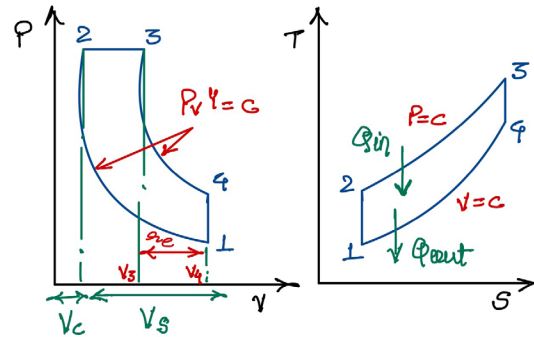
Quality governing
Higher efficiency
Higher compression Ratio (14 to 22)



$$\eta_{th} = 1 - \frac{1}{\alpha_c^{\gamma-1}}$$

Diesel cycle :-

one constant volume
one constant pressure
Two adiabatic process.



Compression Ratio

$$r_c = \frac{V_1}{V_2}$$

Expansion Ratio (r_e)

$$r_e = \frac{V_4}{V_3}$$

Cut-off Ratio

$$P = \frac{V_3}{V_2}$$

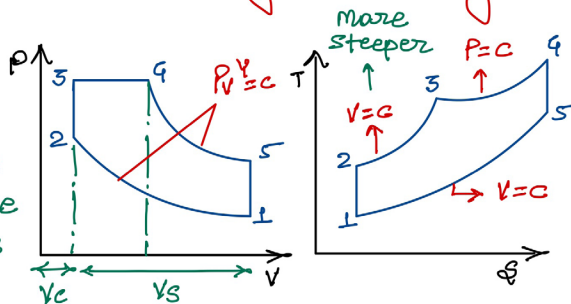
efficiency

$$\eta_{th} = 1 - \frac{1}{\gamma} \left(\frac{1}{r_c^{\gamma-1}} \right) \left(\frac{P^{\gamma-1}}{P-1} \right)$$

As $r_c \uparrow$ or $P \downarrow$ the thermal efficiency of diesel cycle \uparrow .

Dual cycle

Two constant volume
one constant pressure
Two adiabatic process



Pressure Ratio

$$\alpha = \frac{P_3}{P_2}$$

Cut-off ratio

$$P = \frac{V_4}{V_3 = V_2}$$

Efficiency :-

$$\eta_{th} = 1 - \frac{1}{r_c^{\gamma-1}} \left[\frac{\alpha P^{\gamma} - 1}{(\alpha - 1) + \alpha \gamma (P - 1)} \right]$$

When $r_c \uparrow$ or $P \downarrow$ or $\alpha \uparrow \rightarrow \eta_{th} \uparrow$

If $P=1$: Dual cycle becomes otto cycle
If $\alpha=1$: dual cycle becomes diesel cycle.

Comparison among otto, diesel & dual cycles"

Case 1 :- same compression ratio and heat addition

$$\eta_{otto} > \eta_{dual} > \eta_{diesel}$$

Case-2 : same compression ratio & heat rejection

$$\eta_{otto} > \eta_{dual} > \eta_{diesel}$$

In an standard dual air-standard cycle for fixed amount of heat supplied & fixed value of compression ratio, the mean effective pressure shall \uparrow with \uparrow in pressure ratio ' α ' and \downarrow in compression ratio ' ρ '.

otto cycle :- η depends upon volume compression ratio.

diesel cycle :- η depends upon cut-off ratio & volume compression ratio.

Carnot cycle :- η depends upon temp limits

Brayton cycle :- η depends only on pressure limits

Bell Coleman cycle :- Two constant pressure and two isentropic processes

Ericson cycle :- Two constant pressure and two isothermal processes.

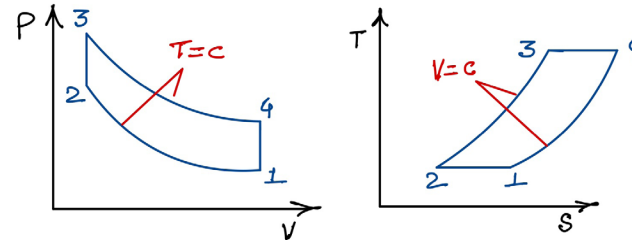
Joule cycle :- Two constant pressure and two adiabatic process.

Stirling cycle :- Two constant volume & two isothermal processes.

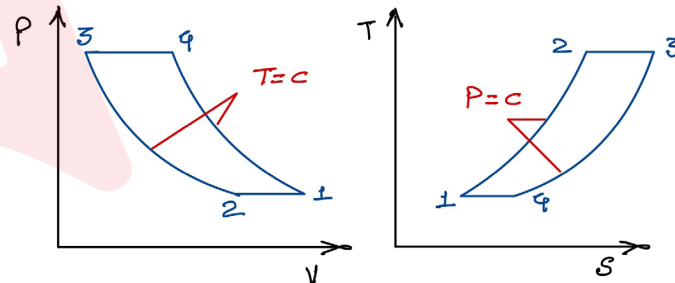
stirling cycle has the highest thermal efficiency for a given max & mini cycle temp.

isothermal compression of air in a stirling engine is an example of closed system with movable boundary.

Atkinson cycle :- Heat supplied at constant volume & rejected at $p=c$.



Stirling cycle



Ericson cycle

Point to Remember



Indicated Thermal efficiency

$$\eta_{I.P.} = \frac{I.P.}{m_f \times CV}$$

$$I.P. = \frac{P_m (LAN) K}{60} \cdot \frac{K}{Z}$$

$$P_m = K \frac{ad}{ld}$$

Brake thermal efficiency

$$\eta_{B.P.} = \frac{b.p.}{m_f \times CV}$$

$$B.P. = \frac{2\pi NT}{60}$$

Mechanical efficiency

$$\eta_{mech} = \frac{B.P.}{I.P.}$$

Volumetric efficiency

$$\eta_v = \frac{\text{Actual volume}}{\text{Swept volume}} = \frac{V_a}{V_s}$$

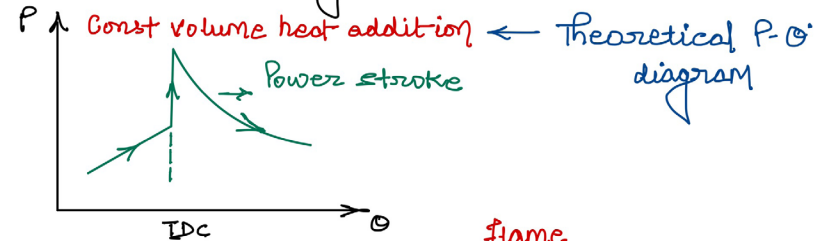
Break specific fuel consumption (bsfc)

$$bsfc = \frac{m_f}{b.p.} \quad \frac{kg}{kwh}$$

Air-fuel Ratio: A/F ratio

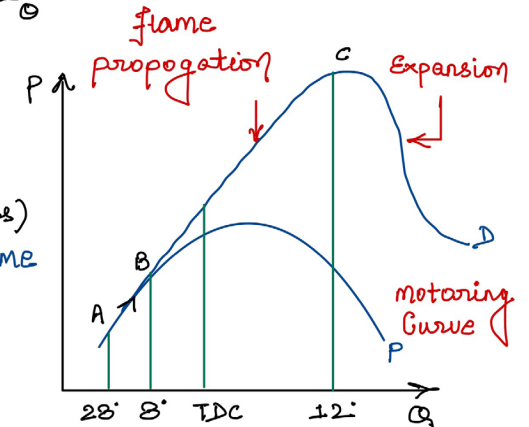
$$A/F \text{ ratio} = \frac{\text{Mass of air } (m_a)}{\text{Mass of fuel } (m_f)}$$

Combustion in S.I. engine



at A: spark initiation
 A-B: Ignition lag
 (Chemical process)
 B-C: propagation of flame
 (Physical process)

C-D: After burning



Ignition lag :- Ignition lag period / prep. phase.
 it is the period in which growth & development of flame takes place.

Power Plant Engineering

Power Plant Engineering

Gas Turbines

Advantage of Gas turbines over IC-engine

- ⇒ Simple mechanism
- ⇒ High speeds are developed: 'Rotary motion'
- ⇒ Easy balancing
- ⇒ Compact in size

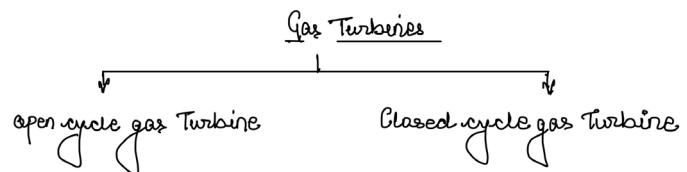
Disadvantages :-

Net output is less as compressor work is high

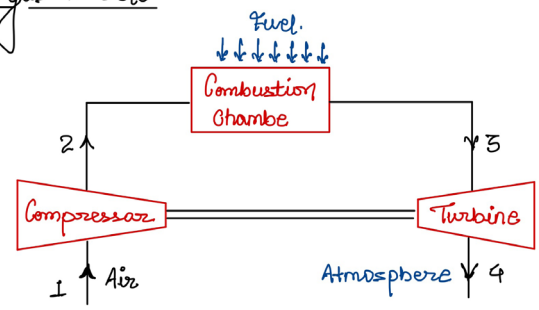
$$W_{net} = W_T - W_C$$

$$\eta = \frac{W_{net}}{Q_s}$$

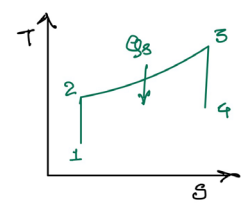
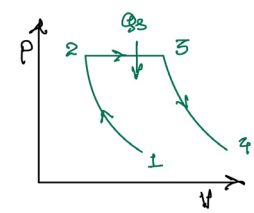
⇒ due to high speeds, reduction gauge are required



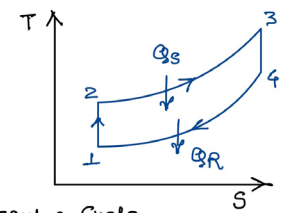
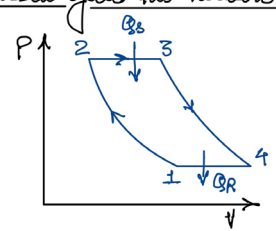
open cycle gas turbine



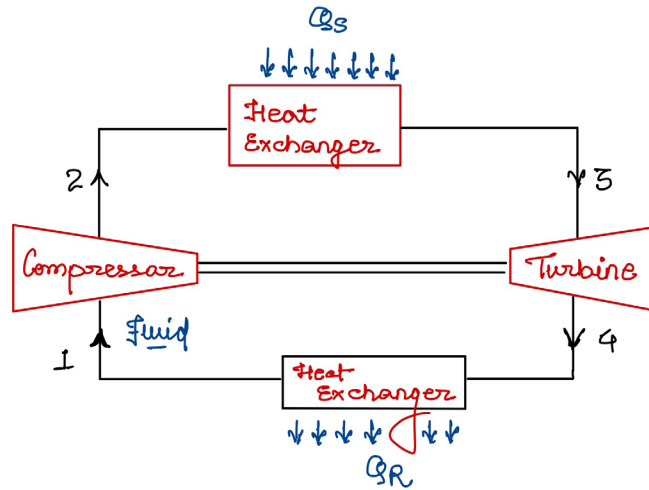
- 1-2 - Reversible adiabatic (isentropic) compression
- 2-3 - Constant pressure heat addition.
- 3-4 - Reversible adiabatic expansion.



Closed Cycle Gas Turbine



Brayton Cycle.



Advantages

* wide variety of fuels can be used, because the product of combustion do not enter the turbine blades.

Disadvantages :-

* the system is complicated & costly
 * coolant is req. for cooling the turbine exhaust, before entering the compressor.

Gas turbine cycles works on Brayton cycle

Work Ratio (η_{wr})

It is the ratio of net work to the positive work.

$$\eta_{wr} = \frac{\text{Net Work}}{\text{+ve Work}} = \frac{W_T - W_C}{W_T}$$

Back work Ratio :- (η_{bwr})

$$\eta_{bwr} = \frac{-\text{ve Work}}{\text{+ve Work}}$$

$$\eta_{bwr} = \frac{W_C}{W_T} = \frac{\text{Compressor Work}}{\text{Turbine Work}}$$

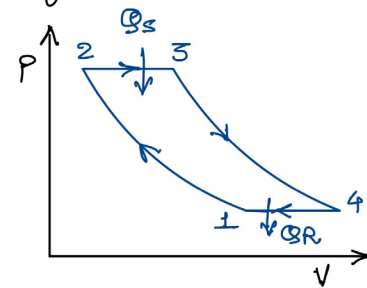
$$\eta_{wr} = 1 - \eta_{bwr}$$

In Rankine cycle, compressor is replaced by pump.

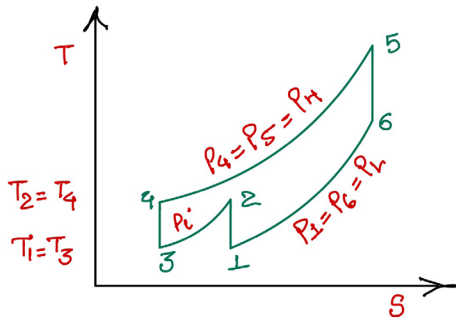
In Gas turbine cycles, the back work ratio is 40-60%
 Where as in Rankine cycle, the back work ratio is 1-2%.

As pump work is negligible in Rankine cycle, the work ratio is almost unity.

Efficiency of a Simple Gas-Turbine cycle



Condition for minimum work input with perfect intercooling.



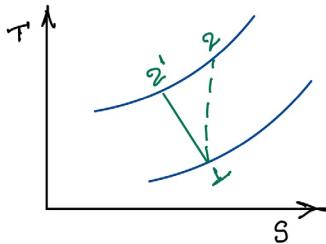
$$T_1 = T_3$$

$$T_2 = T_4$$

$$P_i = \sqrt{P_H \cdot P_L}$$

$$W_{c1} = W_{c2}$$

Isentropic efficiency of air compressor



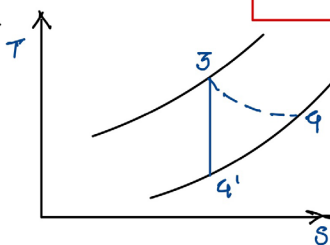
$$\eta_c = \frac{\text{Isentropic Work}}{\text{Actual Work}}$$

$$= \frac{h_{2'} - h_1}{h_2 - h_1}$$

$$= \frac{\gamma_p (T_{2'} - T_1)}{\gamma_p (T_2 - T_1)}$$

$$\eta_c = \frac{T_{2'} - T_1}{T_2 - T_1}$$

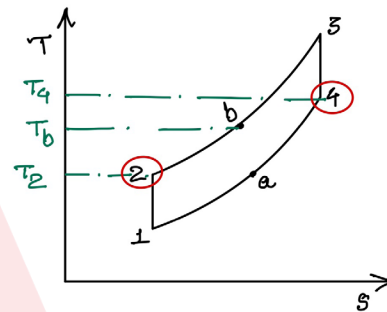
Turbines :-



$$\eta_T = \frac{\text{Actual Work}}{\text{Isentropic Work}}$$

$$= \frac{h_3 - h_4}{h_3 - h_{4'}} = \frac{T_3 - T_4}{T_3 - T_{4'}}$$

Regeneration in Gas Turbines

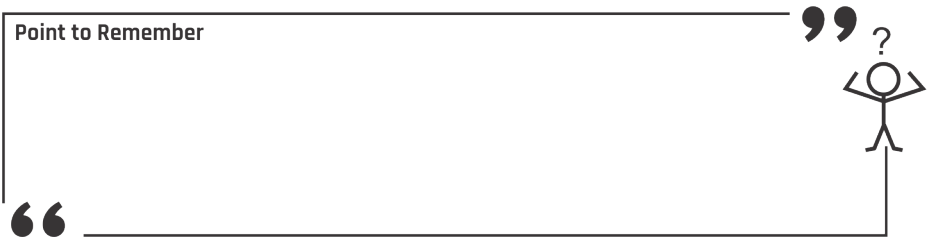


When difference between T_4 & T_2 is large then only we can utilise regenerator upto maximum extent.

Effects of Regenerator

- + No change in compressor work
- + NO change in turbine work
- + No change in net work
- + Reduct (ion) in heat supply
- + Increase in efficiency

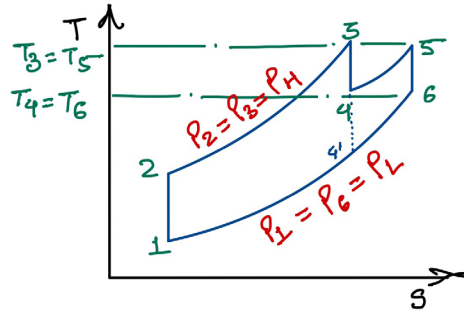
Point to Remember



Effectiveness of Regenerator (ϵ)
or Degree of regenerator.

$$\epsilon = \frac{\text{Actual gain in Temp}}{\text{Ideal gain}} = \frac{t_b - t_2}{t_4 - t_2}$$

Reheating in Gas turbine



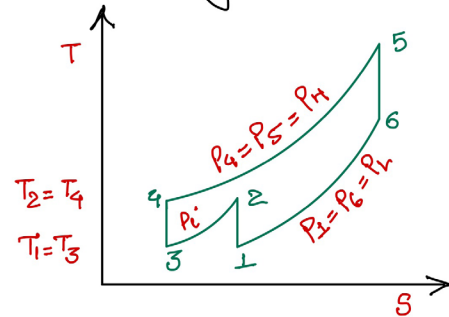
1-2-3-4-5-6-1
Reheat cycle
1-2-3-4'-1
Simple cycle

Effects of Reheating

- * Increase in turbine work
- * No change in compressor work
- * Increase in net work
- * Increase in heat supply

With reheating, the scope for regeneration increases because $T_6 > T_4'$

Intercooling in Gas Turbine



$$W_T - W_C = W_{net}$$

$W_T \rightarrow$ Reheating
 $W_C \rightarrow$ Intercooling

Effects of Intercooling

- * Reduction in compressor work
- * No change in Turbine work
- * Increase in net work
- * Increase in heat supply

With intercooling the scope for regeneration increases & hence intercooling is generally coupled with regeneration.

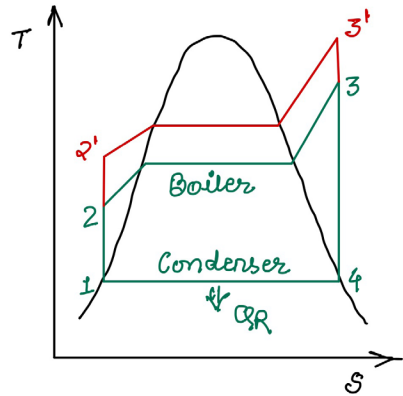
Point to Remember

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By Increasing the Boiler pressure



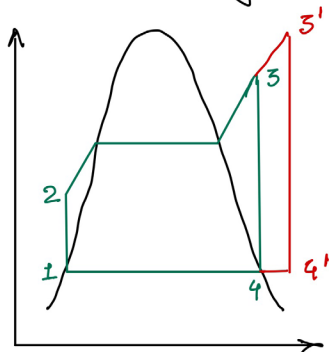
Before 1 2 3 4
After 1 2' 3' 4

$w_T \uparrow$
 $w_P \uparrow$ } $w_{net} \uparrow$

$Q_s \uparrow \downarrow$
 $Q_R \rightarrow$ Constant

$$\eta = \frac{w_{net} \uparrow}{Q_s \uparrow \downarrow}$$

By Superheating of steam :-



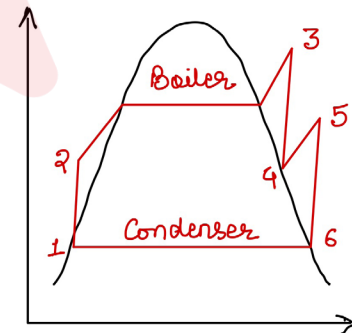
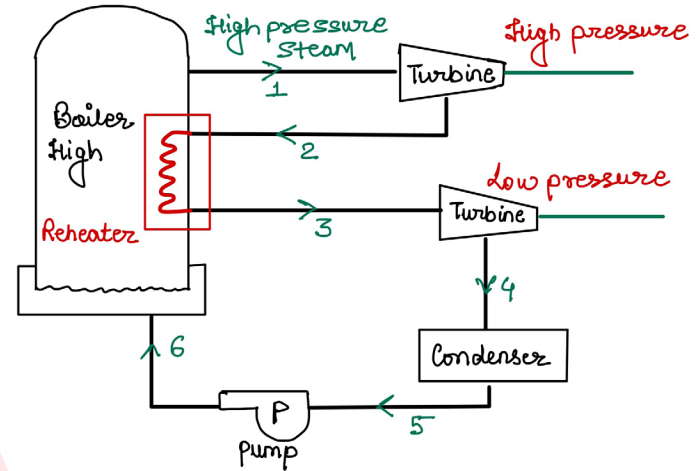
Before :- 1 2 3 4
After :- 1 2 3' 4'

$w_T \uparrow \uparrow$
 w_P constant
 $Q_s \uparrow \uparrow$
 $Q_R \uparrow$

$$\eta = \frac{w_{net} \uparrow \uparrow}{Q_s \uparrow \uparrow}$$

Can't comment on efficiency.

Reheating in Rankine cycle



$w_T \uparrow$
 $w_P =$ Same
 $Q_R \uparrow$
 $Q_s \uparrow$

$$\eta = \frac{w_T - w_P}{Q_s} = \frac{w_T \uparrow}{Q_s \uparrow}$$

Can't comment on efficiency.

Theory of Machines (TOM)

Machine & Mechanism

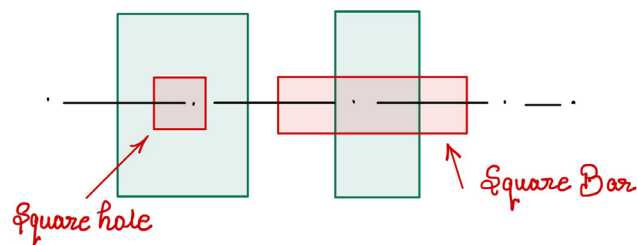
Machine: A machine is a mechanism or a combination of mechanisms which, apart from imparting definite motions to the parts, also transmits and modifies the available mechanical energy into some kind of desired work.

Mechanism: If a number of bodies are assembled in such a way that the motion of one causes constrained and predictable motion to the others, it is known as mechanism.

Types of Constrained motion :-

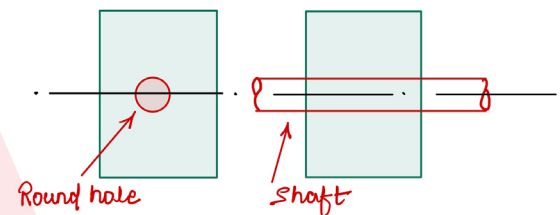
Completely constrained motion

When the motion between the pair is limited to a definite direction, irrespective of the direction of force applied.



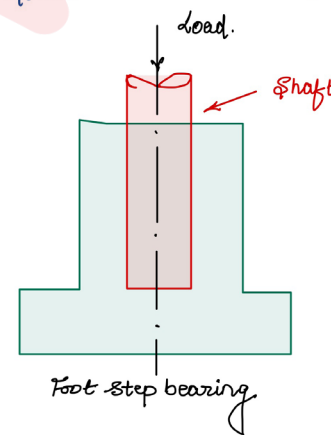
Incompletely constrained motion.

When motion between a pair can take place in more than one direction



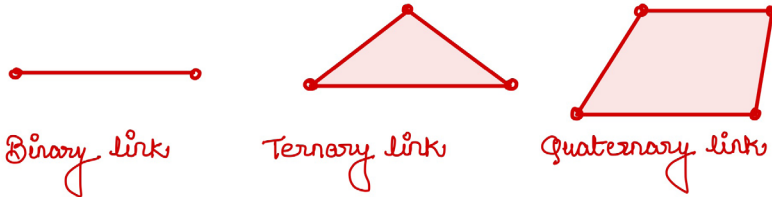
Successfully constrained motion

When the motion b/w the elements, forming a pair, is such that the constrained motion is not completed by itself, but by some other means.



Link: A resistant body or a group of resistant bodies with rigid connections preventing their relative movements is called as link.

Types of links



Kinematic pair When two kinematic links are connected in such a way that their motion is either completely or successfully constrained, then these two links are said to form a kinematic pair.

Classification of kinematic pair

According to nature of contact

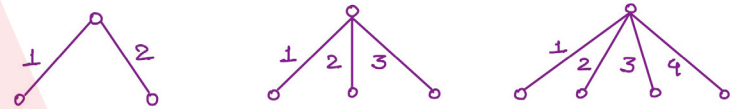
- ↳ lower pair : Area of surface contact
- ↳ Higher pair : Point or line contact
- ↳ Closed pair :- When the elements of the pairs are held together mechanically.
- ↳ Open/unclosed pair :- When the elements of the pairs are in contact either due to force of gravity or some spring action

According to nature of relative motion

- ↳ Sliding pair
- ↳ Turning pair
- ↳ Rolling pair
- ↳ Screw pair
- ↳ Spherical pair

Types of Joints

- # Binary Joint
- # Ternary Joint
- # Quaternary Joint



No. of Binary Joints = $n-1$; $n = \text{no. of links}$

one Ternary Joint = 2 binary Joints

one Quaternary Joint = 3 binary Joints

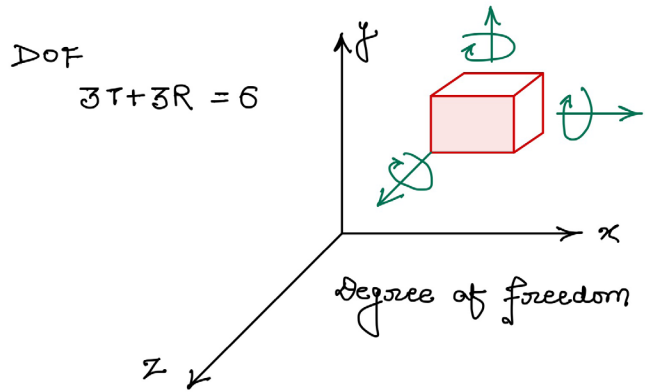
Point to Remember



Degree of freedom

An unconstrained rigid body moving in space can be described by

- a) Translational motion along any three mutually perpendicular axes x, y & z
- b) Rotational motion about these axes.



A rigid body possesses six degrees of freedom in space

Degrees of freedom of a pair = no. of independent relative motions

$DOF = 6 - \text{Number of Restraints}$

All straight line motion have 1 dof.
 Planer motion - 2 DOF
 Projectile motion - 2 DOF
 Rectilinear motion - 1 DOF

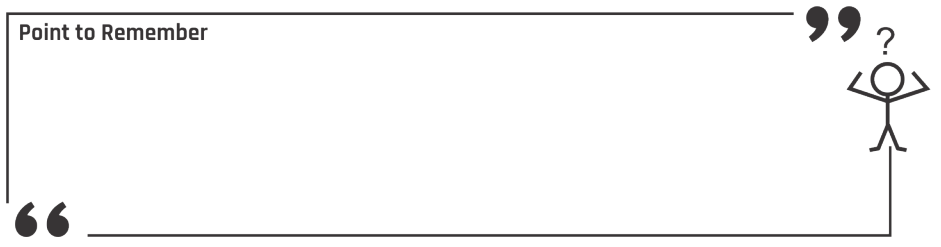
Kinematic chain :- The combination of kinetic pairs in which each links forms a part of two kinematic pairs and the relative motion between the links is either completely constrained or successfully constrained.

for kinematic chain

$N = 2P - 4$; N = NO. of links
 P = NO. of pairs

LHS > RHS → Then the chain is locked
 LHS = RHS → Then the chain is constrained
 LHS < RHS → Then the chain is unconstrained.

Point to Remember

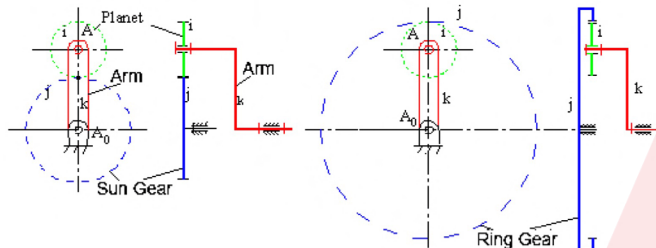


Large speed reductions are possible with epicyclic gears

In general gear trains have two degrees of freedom.

Sun and planet Gear.

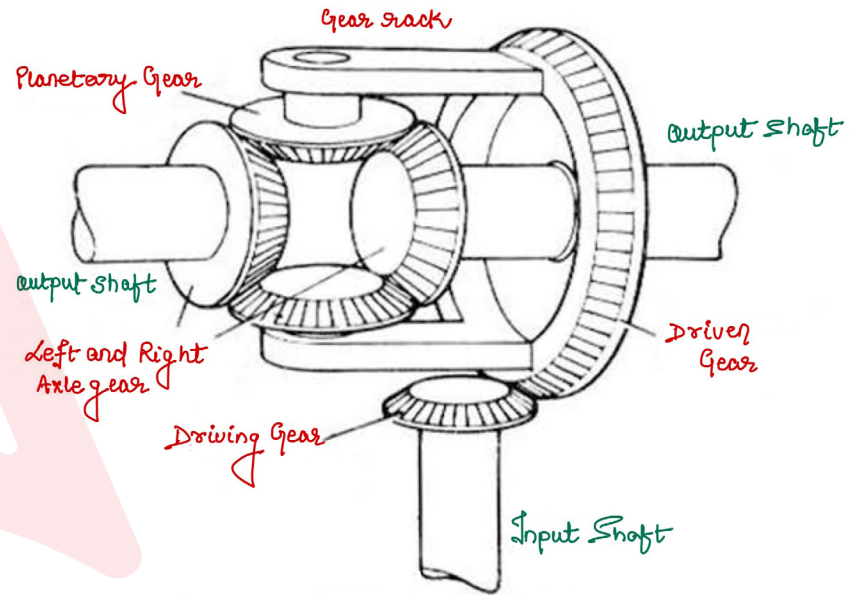
When an annular wheel is added to the epicyclic gear train, the combination is usually referred to as sun and planet gear.



Differential Gear

When the vehicle takes turn, the outer wheels must travel farther than the inner wheels. Since both the rear wheels are driven by the engine through gearing, therefore some sort of automatic device is necessary so that the two rear wheels are driven at slightly different speeds.

This is accomplished by fitting a differential gear on the power (rear) axle. Differential gear is a device which adds or subtracts angular displacements.



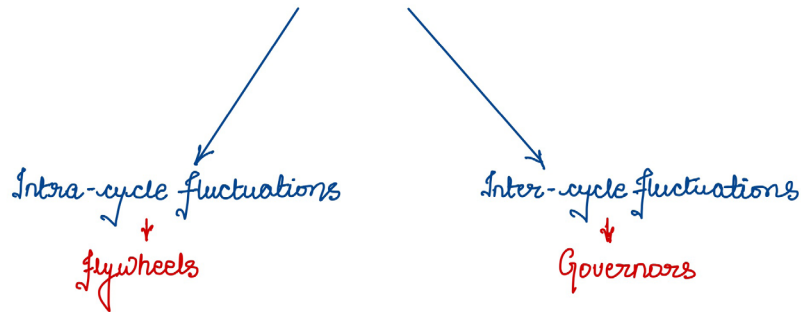
Point to Remember

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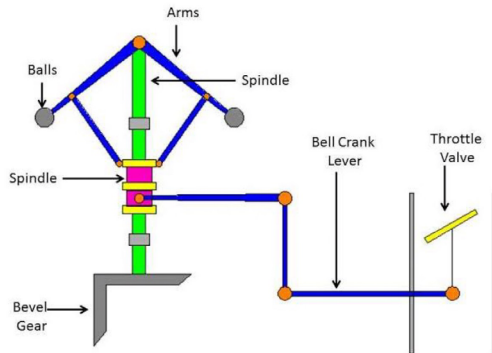
Instantaneous fluctuation of speed control devices



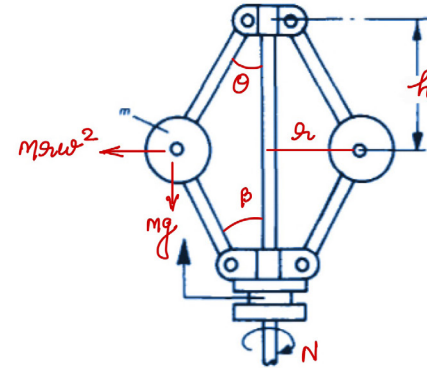
Governors
By Sir James Watt

Basic concepts of centrifugal governors :-

- # When sleeve of governor moves down \rightarrow Throttle opening \uparrow
- # When sleeve of governor hits the bottom stopper \rightarrow Throttle fully open
then engine will develop maximum power.



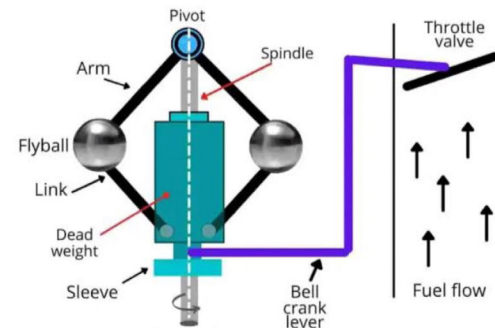
Watt Governor :-



$$N^2 = \frac{895}{h} \quad \text{equilibrium equation of Watt governor}$$

But Watt governor has become insensitive beyond 60 rpm

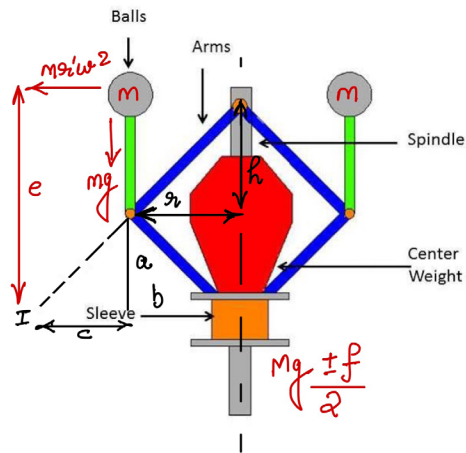
Porter Governor :-



$$N^2 = \frac{895}{h} \left[\frac{2mg + (Mg \pm f)(1+k)}{2mg} \right] \quad K_v = \frac{\tan \beta}{\tan \theta}$$

$$h = \frac{895}{N^2} \left(1 + \frac{(Mg \pm f)(1+k)}{2mg} \right) \quad M \gg m$$

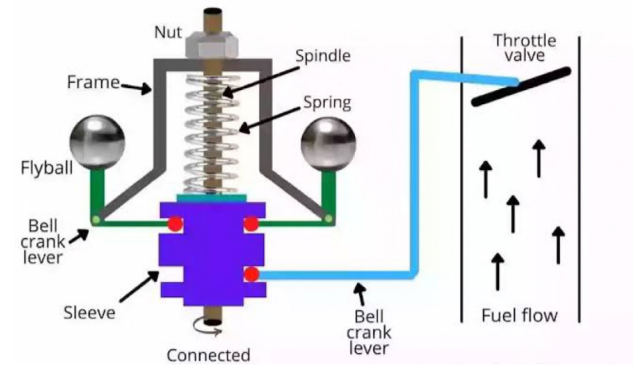
Proell Governor :-



$$(m\omega^2 r) \cdot e = mg(c + r - r') + \left[\frac{Mg \pm f}{2} \right] (b + c)$$

equilibrium eqn of proell governor.

Hartnell Governor
Spring controlled type



Sensitiveness of Governor
when it readily responds to a small change of speed.

$$\text{Sensitiveness} = \frac{\text{Range of speed}}{\text{mean speed}} = \frac{N_2 - N_1}{N}$$

$$= \frac{2(N_2 - N_1)}{N_1 + N_2}$$

N = mean speed
 N_1 = N minimum
 N_2 = N maximum

Metal Cutting

Metal Cutting

Machining Process :- Machining is a manufacturing process in which a sharp cutting tool is used to cut away material to get the desired part shape.

Fabrication vs **Manufacturing**
 Casting, Welding vs M/cing process.

Machine tool :- A machine tool is a machine for handling or machining metal or other rigid materials, usually by cutting, boring, grinding, shearing, or other forms of deformation.

Machine tools employ some sort of tool that does the cutting or shaping.

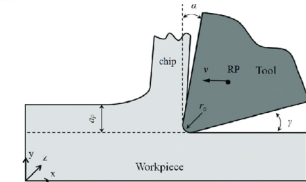
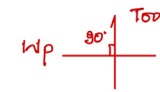
The cutting edge serves to separate a chip from the parent work material.

Cutting tools are classified into two major groups

- # Single point cutting tools
- # Multipoint cutting tools

Orthogonal machining :-

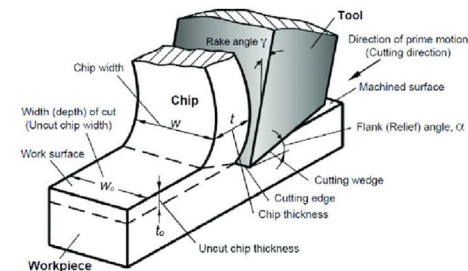
Simplified 2D model of machining that describes the mechanics of machining fairly accurately. It uses wedge-shaped tool, cutting edge normal to direction of cutting speed. Chip is formed by shear deformation along shear plane, oriented at angle ϕ with surface of work.



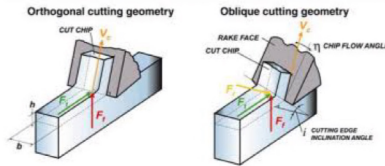
The angle b/w workpiece and cutting tool will be 90°

2-Dimensional Machining.

Cutting edge is wider than the workpiece. The chip generated flows on the rake face of the tool with chip velocity perpendicular to the cutting edge.



Orthogonal vs oblique cutting



Orthogonal cutting

2-Dimensional cutting

Greater amount of cutting force (Constant)

chip flows over the tool face

Relatively shorter tool life

$\phi = 90^\circ$

oblique cutting

3-Dimensional cutting

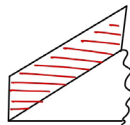
Gradually increasing force

chip flows on the tool face.

longer tool life

$\phi < 90^\circ$

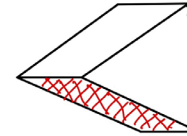
Rake Surface / face



The surface along which the chip moves upward is called face / Rake surface.

Flank / Relief Surface

Flank / Relief



The other surface which is relieved to avoid rubbing with the machined surface is called "Flank" or Relief surface.

Various Angle discussion

Rake Angle :- It allows chip flow direction.

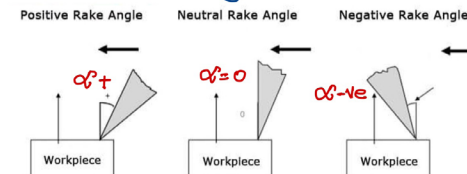
α
It provide keenness (sharpness) to the cutting edge.

It reduces the cutting force required to shear the metal.

It reduce the power consumption
It improves surface finish

Types of Rake angle

- # Positive Rake angle
- # Negative Rake angle
- # Zero Rake angle



Point to Remember



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Casting

Metal casting

Metal casting is a process in which hot liquid metal is poured into a mold that contains a hollow cutout or cavity of the desired finished shape. The liquid metal is then left to solidify, which is removed from the mold, revealing the end product, or the "Casting form".

Pattern :- It is the replica of the part to be cast, (to be produced) and is used to prepare the mould cavity.

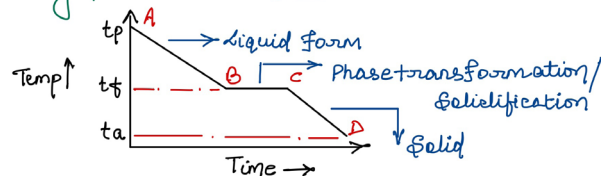
Mould/mold :- It is an assembly of two or more metal blocks, or bonded refractory particles (sand) consisting of a primary cavity.

Tungsten cannot be casted as it is found in powder form, it is produced by powder metallurgy.

Pouring temp = melting temp + ΔT - degree of superheat

Allowances/modifications

Shrinkage/contraction allowance



Liquid & solidification allowance compensated by providing the riser.
Solid shrinkage can be compensated by increasing the size of the pattern.

It is a positive allowance

Shrinkage value

Invar/Bismuth	: Negligible
White metal	: 5mm/m
Cast iron	: 10 mm/m
Aluminium	: 13 mm/m
Copper	: 16 mm/m
Steel	: 20 mm/m
Brass	: 24 mm/m

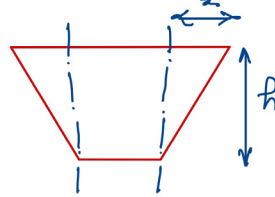
Solid shrinkage is max for Brass
Solidification shrinkage is max for Aluminium
Total shrinkage is max for steel
Grey cast iron will expand when it is cooling.

Ratio of volume of pattern to the casting will be less than one for Grey cast iron and will be more than one for other materials. It will be equal to 1 for Invar & Bismuth.

Draft/Taper allowance :-

The amount of draft allowance will depends on vertical height of the pattern.

$$x = h \tan \theta$$



Machining / Finishing allowance

Expressed in mm/surface
 ↳ Positive allowance

Rapping / Shake allowance

Negative allowance

Distortion / Camber allowance

This is provided opposite to the direction of distortion. This value will depends on lift ratio.

Wax as a pattern : Investment casting (for small objects)

Mercury as a pattern : Mercastr process (-38°C freezing temp)

Evaporative pattern casting - full moulding

Types of pattern :-

Solid/Single piece pattern

very simple in shape & size.

Double piece pattern

Multi piece pattern

Split piece pattern

having complex shapes
 split into (no. of pieces along the parting line)
 getting elements are produced manually

Loose piece pattern

having projections and undercuts
 loose pieces are remove through the cavity by using lifter.

Gated pattern

used for mass production

Match plate pattern

Complex shape of objects in mass production, similar to split piece pattern.

Cope (upper) & drag (lower) pattern

↳ unsymmetric & large size objects

Point to Remember



66

Solidification time :-


Chvorinov's principle

$$t_s \propto \left(\frac{V}{S.A.}\right)^2$$

$$t_s = K_s \left(\frac{V}{A}\right)^2$$


$K_s = \text{Solidification factor } (S/m^2)$

Cube :



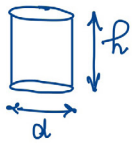
$$\Rightarrow \frac{V}{A} = \frac{a^3}{6a^2} = \frac{a}{6}$$

Sphere



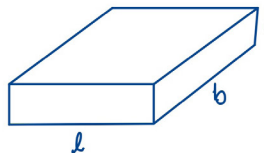
$$\Rightarrow \frac{V}{A} = \frac{\frac{4}{3}\pi R^3}{4\pi R^2} = \frac{R}{3} = \frac{D}{6}$$

Cylinder



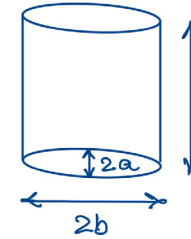
$$\Rightarrow \frac{V}{A} = \frac{\frac{\pi}{4} d^2 h}{2 \cdot \frac{\pi}{4} d^2 + \pi d h}$$

Slab =



$$\Rightarrow \frac{V}{A} = \frac{lbh}{2(lb + bh + hl)}$$

Elliptical



$$\Rightarrow \frac{V}{A} = \frac{\pi \cdot ab \cdot h}{2\pi ab + 2\pi \sqrt{\frac{a^2+b^2}{2}} \cdot h}$$

Riser: It is used to compensate the liquid & solidification shrinkage

cylindrical shape riser is mostly preferred.

$\frac{A}{V} \rightarrow$ cooling characteristics

Types of Riser

- ↳ Side riser
- ↳ Top riser
- ↳ Blind riser

Just a concept not practically used.

Methods to increase performance of Riser

- Providing insulating material on riser surface
- Providing exo-thermic material on top surface
- Providing blind risers
- using optimum condition of designing of riser

Chills method

$$\text{freezing ratio} = \frac{\left(\frac{A}{V}\right)_c}{\left(\frac{A}{V}\right)_R}$$

(FR > 1)

$$x = \frac{a}{y-b} + c$$

$x = \text{freezing ratio}$
 $y = \frac{V_c}{V_r}$, $a, b, c \rightarrow \text{Const.}$

using this method simple (dimensions of sizes) can be calculated in simple shape casting.

Chills and Padding

Chills: To maximise the heat transfer rate and to provide uniform solidification

Padding: To minimise the erosion and to maximise the heat transfer rate.

For chills and padding, metallic objects with high melting point and high thermal conductivity are provided.

Classification of casting techniques

Expandable moulding
(Temporary mould)

- a) Sand moulding
- b) Shell moulding
- c) Investment casting
- d) Full moulding
- e) CO₂ moulding

Permanent moulding
(Metallic moulds)

- a) centrifugal casting
- b) die casting
- c) slush casting

Continuous casting
(Casting + forming)

Shell moulding (Semi-precision casting technique)

Moulding material

fine grain silica
phenolic Resins (Binders)

used for both ferrous and non-ferrous material.

Application :- cylinder heads/blocks of IC engine.
Rocker arms
Valve plates of Refrigerator

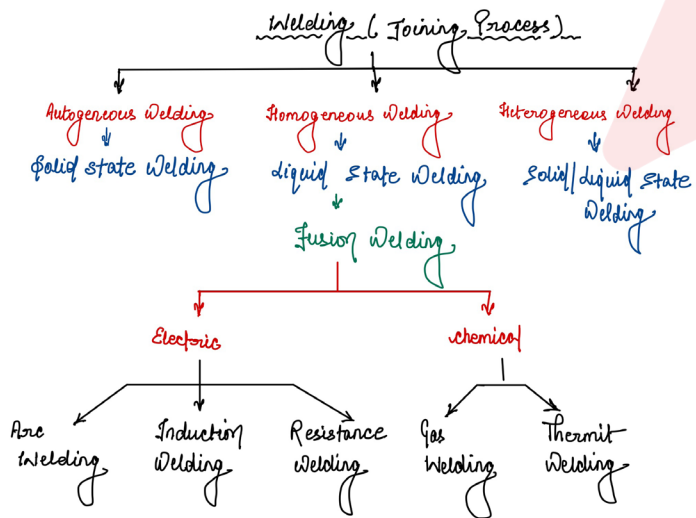
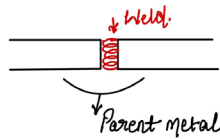
Point to Remember



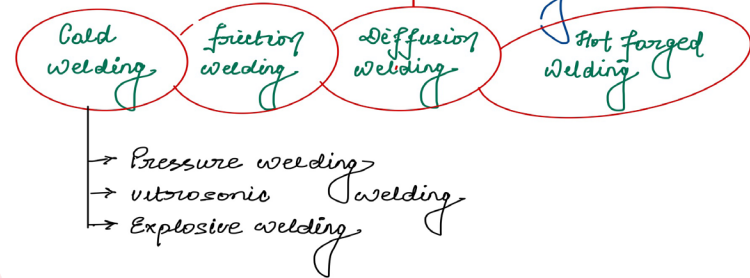
66

Welding

Metal Welding
 Welding is a process of joining together two pieces of metal so that bonding takes place at their original boundary surfaces.
 When two parts to be joined are melted together, heat or pressure or both is applied and with or without added metal for formation of metallic bond.



Solid state welding



Solid/Liquid state welding



Principle of Arc welding

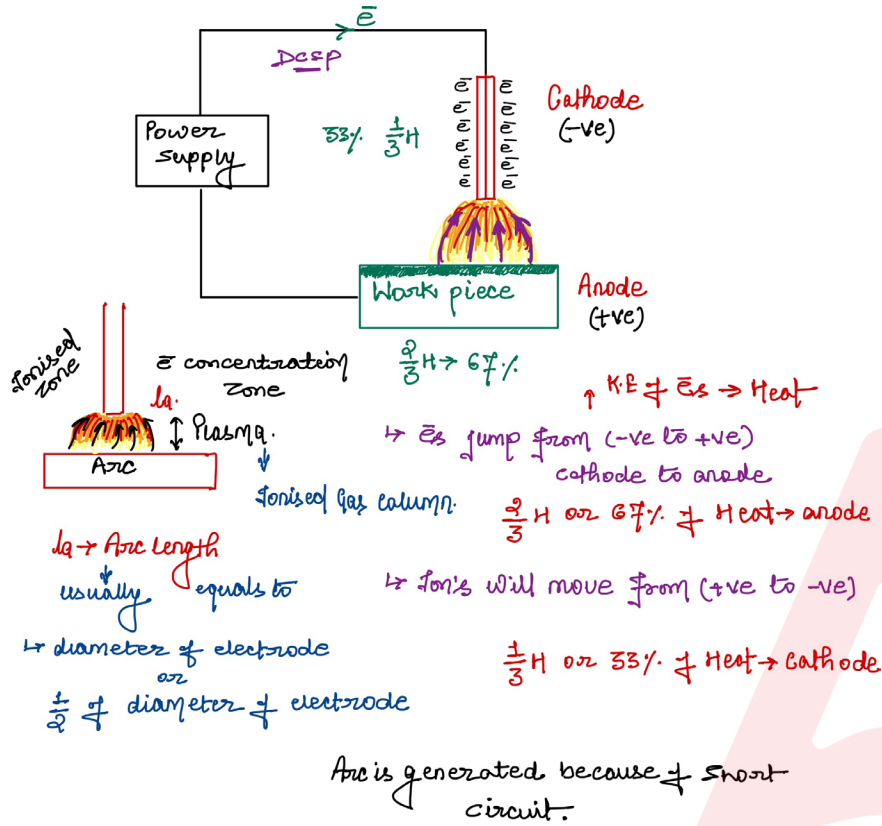
An arc is generated between cathode and anode. When they are touched to establish the flow of current and then separated by a small distance.

65% to 75% heat is generated at the anode.

While using the DC (Direct current)

If the work is positive (the anode of the circuit) the condition is known as "Straight polarity."

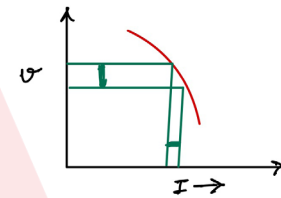
If the work is negative and electrode is positive, this condition is known as "Reverse Polarity."



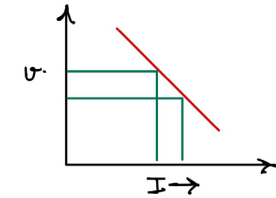
Reverse polarity :- Direct current electrode positive

Workpiece = -ve (DCRP)
Electrode = +ve (DCEP)

	DCRP	DCEP
# Weld Penetration	Shallow	Deep
# Metal deposition	High	Low
# Workpiece thickness	Thin	Thick
# Arc blow	Severe	Severe
# Heat Generation	$\frac{2}{3}H \rightarrow$ electrode	$\frac{2}{3}H \rightarrow$ work piece
	$\frac{1}{3}H \rightarrow$ work piece	$\frac{1}{3}H \rightarrow$ electrode.

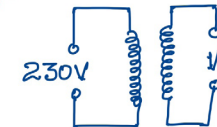


Constant current type
Manual arc welding technique



Constant voltage type
Automatic welding technique.

open circuit voltage (V_0) The maximum rated voltage b/w the terminals under no loading condition.



80-90 Volts

Straight polarity :- Direct current electrode negative

Work piece +ve (DCEP)
electrode -ve (DCEN)

Point to Remember

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ii) Short circuit current - I_s It is a maximum rated current that a welding m/c is capable of supplying the voltage.

Duty cycle :- It is the percentage of time during arc is on without over heating of any component in welding m/c, whenever welding take place. the current passes through the components & produces heat, which may result in overheating, and to avoid this we have to stop the welding process.
Usually 60% duty cycle is the standard industrial rating.

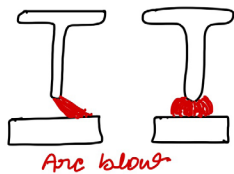
$$\text{Duty cycle} = \frac{\text{Arc on time (AOT)}}{\text{Total welding Time}}$$

Concept of effect of arc blow

Low heat penetration
Excessive weld spatter
Pinch effect in welding is the result of electromagnetic forces.

Weld spatter occurs due to

High welding current
Too small an electrode arc



The effect of arc blow can be minimised with D.c. welding by :-

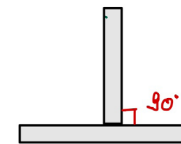
- * Shortening the arc
- * Reduce current
- * Reducing weld speed
- * Balance magnetic field by placing one ground lead at each end of the w/p.
- * Wrapping the electrode cable a few turns around the workpiece.

Common Joint configurations

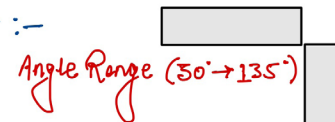
Butt Joint :- Two parts making an angle of $135^\circ - 180^\circ$ inclusive in the region of the joint.



T Joint :-

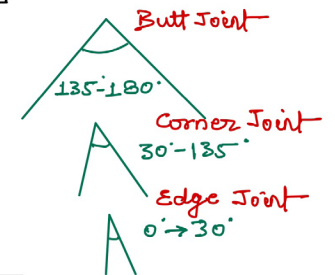
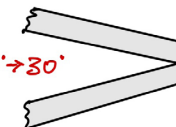


Corner Joint :-



Edge Joint

Angle Range. $0^\circ \rightarrow 30^\circ$



Metrology

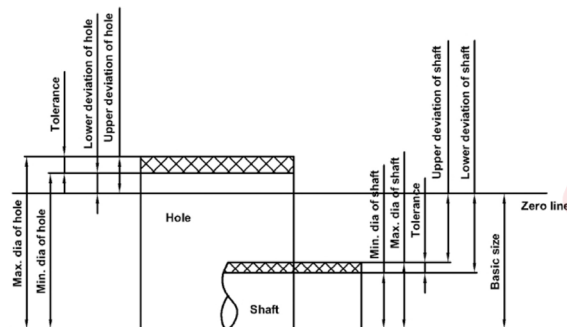
METROLOGY (Science of measurement)

Ex:- Dimensions, temp & forces.

Limit, fit & tolerance.

Shaft :- Anything considered according to the outer diameter

Hole :- Anything considered according to inner diameter



- Tolerance**
- Diff between upper limit & lower limit
 - Maximum possible variation in dimension
 - it may be unilateral / bilateral.

Ex:-

$\phi 25$	+0.18	Basic Size : 25.00MM : 25.00MM
	+0.10	upper limit : 25.18MM : 24.90MM
	-0.10	lower limit : 25.10MM : 24.80MM
$\phi 25$	-0.20	Tolerance : 0.08MM : 0.10MM.

upper deviation : Algebraic diff b/w the maximum size and basic size

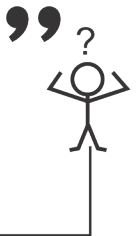
lower deviation : Algebraic diff b/w the minimum size and basic size

Mean deviation : Arithmetical mean of upper and lower deviation

fundamental deviation : It is the deviation of either the upper or the lower deviation, which is nearest one to zero line for either a hole or shaft.

Tolerance can never be zero or -ve.

Point to Remember



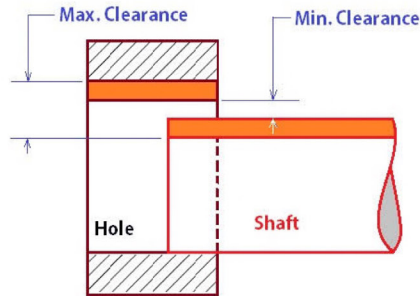
FITS

(Assembly condition b/w 'hole' & 'shaft')

Clearance fit

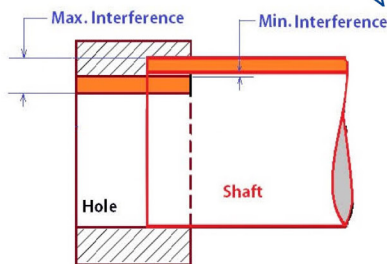
(slide fit, easy sliding, slack running, loose running fit)

$G_{max} = UL \text{ of hole} - LL \text{ of shaft}$
 $G_{min} = LL \text{ of hole} - UL \text{ of shaft}$



Interference fit

(shrink fit, heavy drive fit)



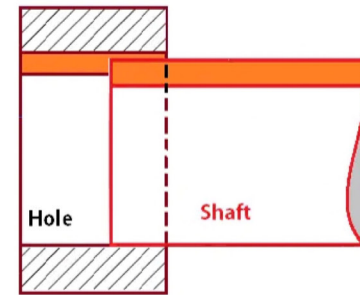
$I_{max} = UL \text{ of hole} - UL \text{ of shaft}$
 $I_{min} = UL \text{ of hole} - LL \text{ of shaft}$

$I_{max} = \text{mini clearance}$
 $I_{min} = \text{Max clearance}$

Transition fit

Mix/Combination of clearance fit & interference fit

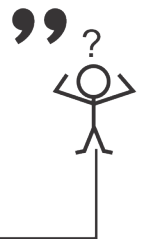
(Push fit, press fit & wringing fit etc)



Allowance (mini clearance or max interference)

It is the intentional difference b/w the basic dimensions of the mating parts, may be +ve/-ve

Point to Remember



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CHAPTER
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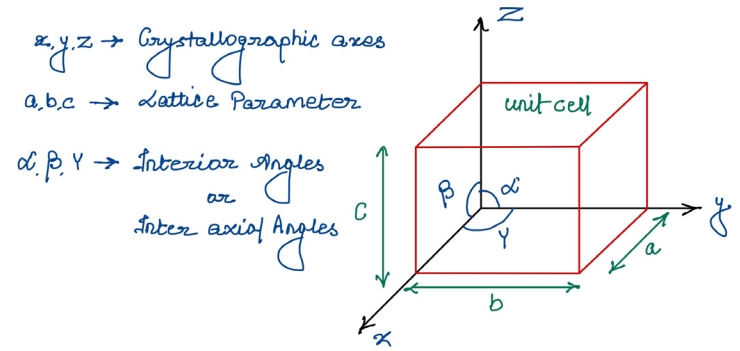
Material Science

Material Science

Crystal structure of unknown materials → x-Ray diffraction Technique

<u>Atom</u>	<u>Ions</u>	<u>Molecules</u>
Atomic Solid Metals	Ionic Solid Ceramics	molecular Solid Polymer (Crystalline Polymer)

Mirror - Glass - Solid - Amorphous → Super cooled liquid
↓
All the polymers are amorphous, but it can be converted into crystalline polymers
↓
are developed only for load bearing capacity.



Crystal System

- Cubic
- Tetrahedral
- Orthorhombic
- Rhombohedral
- Hexagonal
- monoclinic
- Triclinic

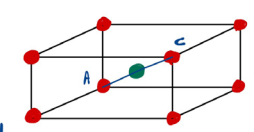
Geometry

- $a=b=c$ & $\alpha=\beta=\gamma=90^\circ$
- $a=b \neq c$ & $\alpha=\beta=\gamma=90^\circ$
- $a \neq b \neq c$ & $\alpha=\beta=\gamma=90^\circ$
- $a=b=c$ & $\alpha=\beta=\gamma \neq 90^\circ$
- $a=b \neq c$ & $\alpha=\beta=90^\circ, \gamma=120^\circ$
- $a \neq b \neq c$ & $\alpha=\gamma=90^\circ \neq \beta$
- $a \neq b \neq c$ & $\alpha \neq \gamma \neq \beta \neq 90^\circ$

Crystal structure characteristics

Lattice Parameter	BCC	FCC	HCP
↓ Atomic Radius			
a to r relation	$a = \frac{4r}{\sqrt{3}}$	$a = \frac{4r}{\sqrt{2}}$	$a = 2r$
Avg no. of atoms	2	4	6
Co-ordination no.	8	12	12
Atomic Packing factor	0.68	0.74	0.74

BCC :-
Hard & Brittle Material



$a \sqrt{3} = 4r$
 $a = \frac{4}{\sqrt{3}} r$

Fe (except in temp range of $910^\circ\text{C} - 1400^\circ\text{C}$)
W, Cr, V, Mo, etc